STUDY OF INDUCTION MOTOR WITH ASYMMETRIC ROTOR

A Dissertation

submitted in partial fulfilment of the requirements for the Degree

of

MASTER OF ENGINEERING

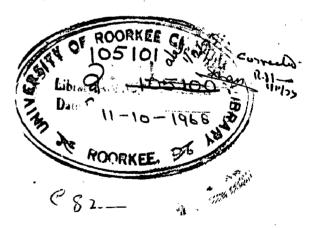
in

ADVANCED ELECTRICAL MACHINES

By.

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DEPARTMENT OF ELECTRICAL ENGINEERING UNIVERSITY OF ROORKEE ROORKEE July, 1968

CREEIFICATE

Cortified that the dissortation entitled 'STUDY OF INDUCTION MOTOR WITH ASYMMETRIC ROTOR' which is being submitted by Sri T.K. Chatterjee in partial fulfilment for the award of the degree of Master of Engineering in Advanced Electrical Machines of the University of Roorkee, Roorkee, is a record of students' own work carried out by him under my supervision and guidance. The matter embodied in this dissertation has not been submitted for the award of any other degree or diploma.

This is further to certify that he has worked for a period of soven months from January 1968 to July 1968, for preparing this discortation for Master of Engineering at this University.

Dated, July 3| 1963.

Brunchspadhyay

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(T.K. Chatterjee)

A_B_S_T_R_A_C_T.

In the present work an attempt has been made to start the single phase induction motor without the starting winding, by introducing solidancy in the rotor. This magnetic asymmetry in the rotor circuit has been investigated in detail and an analysis is made on the basis of two-axis theory. The parameters of importance have been derived and experimentally varified. Further the case of mechanical transients has been studied with a due consideration to non-linearity of torqueopeed curve with the help of computer.

An induction motor with asymmetrical rotor has been designed, fabricated and tested for the above purpose. C_O_II_T_B_II_T_S:

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LIGT OF GUIBOLS

T	o symbolo dofinod bolow have been used vithout
dofinin	these They have the meaning bolous
V _O	Instantonoous applied voltage
\∆ <mark>0</mark>	Instantanoous deauls voltago
₽ ^Q	Instantanoous q=anis voltago.
۲ <u>–</u> ۲	Maximum value of instantaneous applied voltage.
Vel	Instantaneous decuis clip frequency voltage.
V82 1	Instantancous dearso (3-8) frequency voltage.
20 v	Instantancous q=anis slip frequency voltage
v ^g 2	Instantaneous q-anis (2-s) frequency voltage,
8 1	Glip in poue
p	år oporator.
\$ I	Spoed of the rotor.
Yaz	Flux linkago in the d-anis of the/dircult for elly frequency voltage.
402 '	Flux linkago in the dearies of the stator circuit for (2-0) frequency voltage.
491	Flux linkago in the geamle of the stator elecuit for slip frequency voltage.
Yge '	Flux Linkago in the q-anio of the stator circuit for (2-a) frequency voltage.
Lo	Stoady-component of colf-inductance of stator elreult.
ro .	of statop Doublo frequency component of solf-inductorso/circuit
Er '	Voltago aeroso the rotor elreuit in the Gearle at elly frequency.
_2 2075	Voltago across the rotor circult in the deards at (2-0) frequency.

- $\nabla_{q1}^2 = \text{Voltage across the rotor circuit in the q-axis at slip frequency.}$
- $\nabla_{q2}^{F} \simeq \text{Voltage across the rotor circuit in the q-anis at (2-2) frequency.}$
- Yrdl " Flux linkago of the rotor circuit in the deaxis for slip frequency voltage.
- Yrd2 " Plum linkago of the rotor circuit in the dennis for (2-5) frequency voltage.
- Yrgl Flux linkago of the rotor circuit in the q-amis for slip frequency voltage.
- $\Psi_{rq2} = Plux-linkago of the rotor circuit in the q-axis for (2-0) frequency voltage,$
- Lad "Hutual inductance between the stator and the rotor in the deaxis.
- Lag = Mutual inductance between the stater and the rotor in the graxis.
- L_{red} = Solf-inductance of the rotor circuit in the d-axis.
- L_{rd} = Solf-inductance of the rotor circuit in the q-axis.

Lo = Self-inductance of the d-anis stator circuit.

L_a = Solf-inductance of the q-anis stator circuit.

L_A(n)= Operational inductance of the deaxic stator circuit.

 $L_n(p) = O$ porational inductance of the q-axis stater circuit.

 $\nabla_{\gamma} = R.H.S.$ value of slip frequency voltage.

V₂ = R.H.S. value of (2-5) frequency voltage.

X₁(p)= Operational recotance in the d-axis stator elecuit.

X_q(p)= Operational reactance in the q-anis stator circuit.

 $\mathbf{P}_{\mathbf{n}}$ = Resistance of the stator circuit.

Eq1 = R.M.S. value of d-axis stator circuit at slip frequency.

I

	I _{as}	= R.H.S. value of doamle stator current at (2-s) frequency.
	Igl	s RolleS. value of quarks stator current at ally frequency.
	Sp ^I	= R.M.S. value of q-axis stator current at (2-3) frequency.
•	Ia	= R.M.S. value of the q-axis stator current
	Iq	= R.N.S. value of the q-anio stator current.
t	Ig	= R.M.S. value of stator current.
	۹0	= Instantancous value of stator current,
	¹ n1	= Instantaneous maximum value of d-axis stator current at slip frequency.
	¹ n2	= Instantoneous maximum value of q-axis stator current at slip frequency.
	1 ₈₃	 Instantaneous maximum value of deaxis states current at (2-s) frequency.
	1-04	Instantaneous maximum value of q-axis stator current at (2-s) frequency.
	P _o	= Electro-magnetic developed torque
	Pn	= Averege Botor-torquo.
	8	= Instantaneous value of torque.
	"l	= Loakage roactance of stator circuit
	P _P d	- Resistance of the rotor circuit in the donnis,
	^R rd	- Loekago reactance of the rotor circuit in the dearis.
	r _{PQ}	= Resistance of the rotor circuit in the q-axis,
	D.S.	= Leahage reactance of the rotor circuit in the geaule.

CHAPTER-I

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INTRODUCTION:

Hany-fold applications of single phase induction motors in domestic and industrial fields have drawn the attention of several authors. But all of them have been confronted with the well-known difficulty of single phase induction motor that it lacks inherent starting torque. This is obvious from the fig.l.l which shows a normal cage rotor excited by a single stator winding. The current flowing in each of cage bar reacts with the gap flux producing a torque. The not torque averaged over the entire periphery is zero. In other words the machine is symmetric about the neutral axis NN'. To start the motor, this symmetry should be disturbed. This can be done by introducing asymmetry either in the electric or in the magnetic circuit of the motor. The single phase induction motors in use to the method of starting, aro

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- 1) Split phase motors
- 11) Capacitor motor
- 111) Shaded pole motor.

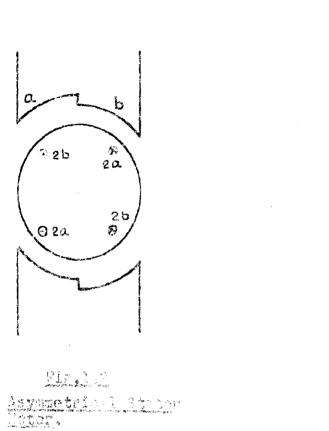
In the first two cases, the electrical asymmetry is introduced by connecting an auxiliary winding in the stator in the stator besides the main winding. The two stator windings are of different impedances and hence they cary currents of different magnitudes and phases. The auxiliary winding is normally placed in space quadrature to the main winding. The time and space phase differences between the two winding m.m.fs. give rise to a resultant field having a rotating component. The squirrel cage rotor reacte with the rotating field to produce starting torque.

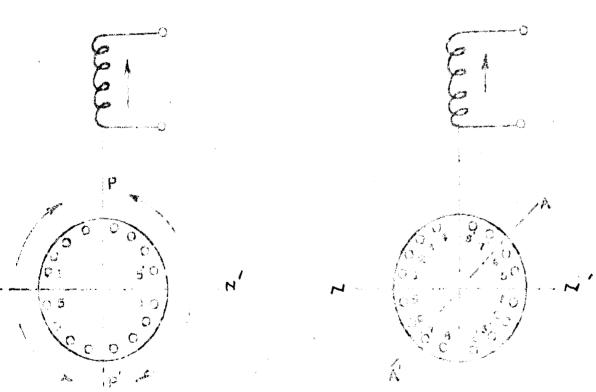
In the split phase type, the auxiliary vinding is of finer vire gauge and hence a higher resistance than the main vinding. This results in a time phase difference between the currents in the two vindings. In the capacitor motor, a capacitor is connected in cories with the auxiliary winding. The time phase spliting is more in the case of capacitor motor than the split phase one and hence the starting torque.

In the third case, the magnetic asymmetry is introduced by imparting caliency in the stater structure. Around oneportion of each pole is wrapped a copper strip, forming a closed circuit. This 'shading coil' dolays the flux passing through it, so that the flux lags in phase bohind that in the unshaded part. This gives rise to a sweeping action, magnetically, across the face of the pole, resulting a revolving flux and thereby the starting torque.

Besides the above three types, single phase repulsion motors are also used. The repulsion motor has a rotor winding like a d.c. machine armature winding and is connected to a commutator. The stater has a single winding. Brushes are kept short circuited and the brush axis is at a space angle to the stater winding axis. When the stater winding is excited, induced currents flow in the short circuited rotor winding. The rotor winding m.m.f. is at a space angle to the stater winding m.m.f. The two winding m.m.f's react with each other to produce a torque.

A radically different method of starting was suggested by Baun². He has stated ^pA single phase induction motor (Fig.1.2) may be given starting torque by varying the rotor constants over the pole face either by asymmetrical iron structure or windingⁿ. Starting torque arises from the difference in leakage reactances of the rotor bars on both sides of the pole structure. It causes a different current to flow on each side producing a resultant torque. The difference in the mutual reactances on two sides of the poles makes





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<u> 19.1.6</u>

a phase difference between the currents flowing in both sides. This also attributes to the resultant torque.

The methods suggested by him to get the asymmetry are:

1) To wind more turns about one side of pole face than about the other side thus causing difference in the permoability of both the rotor and stator iron.

ii) To suitably shape the stator making the airgap under each pole non-uniform. Noither this method has been taken advantage of in practice nor the exact implications underlying this principle seem to have been thoroughly studied.

The constant loss in the shaded ring has attributed to the lover officioncy of shaded-pole motor. The short-circuited squirrel cage vinding on the rotor itself has the effect of shading. Considering those points Doshpando and Desai³ have suggested a type of single phase induction notor with asymmetrical rotor which they name as 'a single phase induction motor with partially open cage rotor', Fig. (1.3) shows a cago whore some slots have been left open without cage bars. The end-ringo are complete. Nov, refering to Fig. (1.1) it can be shown that the clockwise torque due to bars 1-1' is being cancelled by the counterclockwice torque due to similarly situated bars 5-5' and so on, producing a not zero torque. But in the case of Fig. (1.3) there are no bars in the slots 6-6' and the clockwise torque due to barks 2-2' is left uncompensated and the rotor vill experience a clockwise torque evon when excited by a single stater winding. The axis AA' of the opon slots will try to align with the noutral axis MN'. The magnitude and direction of the torque depend on the position of AA*. However. as the rotor starts rotating under the action of this torque, the torque vill also change in magnitude and direction. When the axis AA' turns past the axis NN. It is obvious that the direction of the torano

is reversed. But the rotor in the meatime will devolop speed voltage and will have a tendency to maintain rotation. Also the rotor will have an angular momentum and store mechanical energy and hence it can not be brought to rest instantaneously by the reversing torque. If the point A is able to go past P', it will get again an accolerating torque.

The absence of bars has introduced an asymmetry in the electrical circuit of the rotor. It may however be noted that the magnitic circuit of the Bobor is absolutely symmetrical. Essai has suggested many ways of creating the asymmetry namely use of different bar materials, use of different bar sections etc.

Another asymmetrical rotor motor has been developed by Subba Rao. He has suggested a double cage rotor with all the inner slots and some of the outer-slots filled with suitable bars connected to the same ond-ring $\operatorname{Fig}_{\sigma}(1_{\sigma}G)$. This is evidently a case of electrical asymmetry of the rotor.

But in the present case, an attempt has been made to exploit the magnetic property (reluctance) of the material to create the asymmetry and thereby to start the single phase induction motor with only one stater coil. The construction of the machine is simplest one. It may seen paradoxical that the motor though the simplest in structure presents great difficulties when analysed quantitatively. This reluctance-start induction motor is different from reluctance motor. It can be explained from the Fig.(1.%). Since the flux passing through the inter-polar zone traverses less iron path in the magnetic circuit compared to the air-gap path. The damping of flux due to eddy currents in the iron is also less. Therefore, the flux passing through the inter-polar portion leads the flux passing through the pole section.

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gap, and the motor follows the flux movement. In other words the rotor-axis will always try to align to the minimum reluctance path. The position of rotor with its polar-axis shifted from the stator axis will always produce some torque to align its polar-axis with the stator axis. Hence it can be started in any direction depending upon its rotor position.

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CHAPTER - II

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2.1. GEOMETRY OF THE MACHINE:

The machine fabricated by the author has a stator of laminated sheet steel with slots. The rotor is dumbell-shaped and hence a salient pole type one. An air-gap separates the rotor from the stator.

The electrical circuits of the machine can be differentiated, into two classes:

1) the stator circuits, and

ii) the rotor circuits.

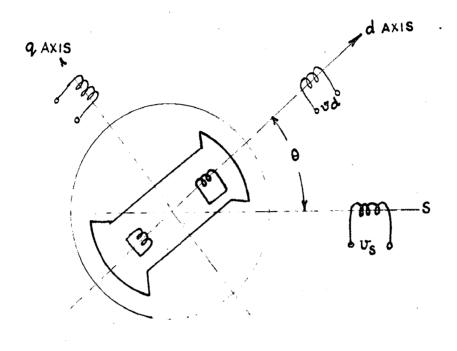
In the stator, each winding may be regarded as constituting an electric circuit. The rotor circuit consists of a no.of copper bars arranged in the form of a squirrel cage with the ends shorted by copper rings. This arrangement of conductors becomes a network and the current distribution can be determined either by the node-pair method or preferably, by the cyclic-current method.

2.2. REFERENCE FRAME AND ASSUMPTIONS:

The rotating machine analysis demands the selection of some reference frame. This reference frame may be attached either to the stator or to the rotor or a reference frame stationary for stator, and rotating for the rotor may be chosen. Besides, it is also possible to select reference frames which rotate at an arbitrary speed not connected with the rotor speed. In the present case, the reference frame is fixed on the rotor and is moving at the rotor speed, > rad/sec.

The machine can be represented by Fig. (2,1). With the reference frame chosen, the machine analysis has been done with the following idealised assumptions.

(



<u>terress retion of the asymptotical</u>

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i) The magnetic paths and all electric circuits are symmetrical about the pole and interpole axis.

11) Hysteresis saturation and other non-linear effects are neglected.

111) The effect of space harmonics in the flux wave is also neglected.

2.3. ANALYSIS:

The stator winding is splitted into two fictitious axis namely d-axis and q-axis.

 $v_{d} = v_{s} \cos \theta$ $v_{q} = v_{s} \sin \theta$ where $\theta = (1-s)$ wt. and $v_{s} = v_{d} \cos \theta - v_{q} \sin \theta$ Assuming $v_{s} = v_{m} \cos vt$. • $v_{d} = v_{m} i \cos vt$. • $v_{d} = v_{m} i \cos vt$. cos (1-s) wti $= \frac{v_{m}}{2} i \cos s wt + \cos (2-s) wt i$ $= v_{d1} + v_{d2}$ when $v_{d1} = \frac{v_{m}}{2} \cos s wt$. $v_{d2} = \frac{v_{m}}{2} \cos (2-s) wt$. Similarly,

> vq = - Vm I cos wt. sin (1-s) wt I = -Vm I sin s wt - sin (2-s) wt I = vq1 + vq2 - Vm/2 sin s wt

9

$$v_{q2} = -\frac{v_m}{2} \sin (2-s) vt.$$

Now, using Park's equation for each frequency voltage, it can be written,

where $p = j_{sw}$, 2 = (1-s)w for the motor running with a constant slip s, 1.e. at steady state.

and for (2-s)w rad/sec. frequency voltage-

$$v_{d2} = p \psi_{d2} + \partial \psi_{q2} + r_s \frac{1}{d2}$$

 $v_{q2} = p \psi_{q2} - \partial \psi_{d2} + r_s \frac{1}{q2}$
 $v_{q2} = p \psi_{q2} - \partial \psi_{d2} + r_s \frac{1}{q2}$

when p = j(2-s)w and $\sqrt[3]{b(1-s)w}$ for steady state operation. Now, it can be shown that-

 $\psi_{d1} = i_{d1}^{s} (L_{o} + \frac{L_{2}}{2}) + L_{ad} i_{d1}^{r}$

 $\psi_{q1} = i_{q1}^{s} (L_{o} - \frac{L_{2}}{2}) + L_{aq} i_{q1}^{r}$ and $\psi_{d2} = i_{d2}^{s} (L_{o} + \frac{L_{2}}{2}) + L_{ad} i_{d2}^{r}$ $\psi_{q2} = i_{q2}^{w} (L_{o} - \frac{L_{2}}{2}) + L_{aq} i_{q2}^{r}$

In the same way, the Park's Equation for the rotor circuits.

$$v_{d1}^{r} = p \not\downarrow_{d1}^{r} + r_{rd} \dot{r}_{d1}$$

 $v_{q1}^{r} = p \not\downarrow_{rq1}^{r} + r_{rq} \dot{r}_{rq1}$
where $p = j_{sv}$ for steady-state operation.
(3)

$$\mathbf{v}_{d2}^{\mathbf{r}} = \mathbf{p} \quad \psi_{\mathbf{r}d2}^{\mathbf{r}} + \mathbf{r}_{\mathbf{r}d} \quad \mathbf{i}_{\mathbf{r}d2} \quad \mathbf{i}_{\mathbf{r}d2} \quad \mathbf{i}_{\mathbf{r}d2} \quad \mathbf{i}_{\mathbf{r}d2} \quad \mathbf{v}_{q2}^{\mathbf{r}} = \mathbf{p} \quad \psi_{\mathbf{r}q2}^{\mathbf{r}} + \mathbf{r}_{\mathbf{r}q} \quad \mathbf{i}_{\mathbf{r}q2} \quad \mathbf{$$

when p = j(2-s)w for steady-state operation. Again, the flux linkages for the rotor circuits can be calculated as follows:

$$\psi_{rd1}^{r} = L_{ad} \, \mathbf{i}_{d1}^{s} + L_{rd} \, \mathbf{i}_{d1}^{r}$$

$$\psi_{rq1}^{r} = L_{aq} \, \mathbf{i}_{q1}^{s} + L_{rq} \, \mathbf{i}_{q1}^{r}$$

$$\psi_{rd2}^{r} = L_{ad} \, \mathbf{i}_{d2}^{s} + L_{rd} \, \mathbf{i}_{d2}^{r}$$

$$\psi_{rq2}^{r} = L_{ad} \, \mathbf{i}_{q2}^{s} + L_{rq} \, \mathbf{i}_{q2}^{r}$$

Putting the values of $\psi_{d1}, \psi_{q1}, \psi_{rq1}, in (1)$ and (3)-

$$v_{d1}^{s} = p \text{ i } i_{d1}^{s} (L_{0} + \frac{L_{0}}{2}) + L_{ad} i_{d1}^{r} \text{ i } + \sqrt{1} i_{q1}^{s} (L_{0} - \frac{L_{2}}{2}) + L_{aq} i_{q1}^{r} \text{ i } + r_{s} i_{d1}^{s}$$

$$v_{q1}^{s} = p \text{ i } i_{q1}^{s} (L_{0} + \frac{L_{2}}{2}) + L_{aq} i_{q1}^{r} \text{ i } - \sqrt{1} i_{d1}^{s} (L_{0} + \frac{L_{2}}{2}) + L_{ad} i_{d1}^{r} \text{ i } + r_{s} i_{q1}^{s}$$

$$v_{d1}^{r} = p \text{ i } i_{d1}^{s} L_{ad} + E_{rd} i_{d1}^{r} \text{ i } + r_{rd} i_{d1}^{r}$$

$$v_{q1}^{r} = p \text{ i } i_{q1}^{s} L_{aq} + L_{rq} i_{q1}^{r} \text{ i } + r_{rq} i_{q1}^{r}$$

$$\dots (5)$$

Writing $L_d = (L_o + L_2/2)$ and $L_q = (L_o - L_2/2)$ and arranging in the matrix form, the following matrix (5) can be developed.

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vå1		(pL _d + r _s)	⇒ r ^d	pLad	⇒ L _{aq}	1 ⁸ dl
v ^s q1	-	- > La	(pL _q + r _s)	-,)Lad	pLaq	igl
val		pLad		(pL _{rd} +r _{rd})		1 ^r a1
yr yq1			pLaq		(plrq+rrq)	irqı

•••(5)

In the same way, writing value of \forall_{d2} , \forall_{q2} , \forall_{d7} , \forall_{q7} ,

 $v_{d2}^{s} = pi \; i_{d2}^{s} \; (L_{0} + \frac{L_{2}}{2}) \pm L_{ad} i_{d2}^{r} i + \frac{1}{2} i_{d2}^{s} (L_{0} - \frac{L_{2}}{2}) L_{ad} \; i_{d2}^{r} i + r_{s} i_{d2}^{s}$ $v_{d2}^{s} = pi \; i_{d2}^{s} \; (L_{0} - \frac{L_{2}}{2}) + L_{ad} i_{d2}^{r} i - \frac{1}{2} i_{d2}^{s} \; (L_{0} + \frac{L_{3}}{2}) + L_{ad} i_{d2}^{r} i + r_{s} i_{d2}^{s}$ $v_{d2}^{r} = pi i_{d2}^{s} \; L_{ad} + E_{rd} \; i_{d2}^{r} i + r_{rd} \; i_{d2}^{r}$ $v_{d2}^{r} = pi i_{d2}^{s} \; L_{ad} + E_{rd} \; i_{d2}^{r} i + r_{rd} \; i_{d2}^{r}$ $v_{d2}^{r} = pi i_{d2}^{s} \; L_{ad} + L_{rd} \; i_{d2}^{r} i + r_{rd} \; i_{d2}^{r}$ $\cdots \quad (6)$

Now writing $L_d = (L_o + L_2/2) \& L_q = (L_o - L_2/2)$ the following matrix(6) can be formed:

v å2		$(pL_d + r_s)$	[,] ⇒ L _q	" pL _{ad} "	SLag '	1 ⁸ 1 ⁸ 12
4 8 ▲		\? La	(pLq+rs)	->Lad	pLaq	i ^s q2
vr 25		pLad		(pL _{rd} +r _{rd})		1 ^r .
A ^{dS}	-	· ·	pLaq		(pL _{rq} +r _{rq})	i ^r q2

• • • • (6)

Now taking the values of $v_{d1}^r = v_{q1}^r = v_{d2}^r = v_{q2}^r = 0$ and reducing the matrices (5) and (6) to the (2 x 2) matrix form and representing the impedance matrix in its operational form (vide Appendix I), the matrices (5) and (6) can be written as (7) and (8).

The matrix (5) reduces to-

$$v_{d1}^{s}$$
 $r_{s} + pL_{d}(p)$ $\lambda L_{q}(p)$ i_{d1}^{s} v_{q1}^{s} $-\lambda L_{d}(p)$ $r_{s}+pL_{q}(p)$ i_{q1}^{s}

where p=]sw, >= (1-s)w

In the same way the matrix (6) reduces to-

$$\mathbf{v}_{d2}^{s}$$
 $\mathbf{r}_{s}^{+pL_{d}}(p)$ $\overset{\backslash}{\sim} L_{q}(p)$ \mathbf{i}_{d2}^{s} $\ldots (8)$ \mathbf{v}_{q2}^{s} $-\overset{\backslash}{\sim} L_{d}(p)$ $\mathbf{r}_{s}^{+p} L_{q}(p)$ \mathbf{i}_{q2}^{s}

where p = j(2-s)w, 2 = (1-s)w

Nov, Let-

 $v_{d1}^{8} = V_{m}/2 \cos swt = V_{1}$ • $v_{q1}^{8} = V_{m}/2 \sin swt = - JV_{1}$ & $v_{d2}^{8} = V_{m}/2 \cos (2-s) wt = V_{2}$ • $v_{q2}^{8} = -V_{m}/2 \sin (2-s) wt = JV_{2}$

Hence for the steady-state operation can be represented by-

v,	r _s +jsX _d (jsw)	(1-s)X _q (jsw)	I _{d1}	,
-jv ₁	-(1-s)X _d (jsw)	$r_s + jsX_q(jsw)$	Iql	(9)

₹2	r _s +j(2-s)X _d Ij(2-s)w	(1-s)XqI\$(2-s)vi	Ids
j⊽ ₂	-(1-3)Xd11(5-3)VI	r_+1(2-5)X_11(2-4)v	I _{q2}

... (10)

Now from (9)-

$$I_{d1} = \frac{V_1 \ i \ r_s + j X_q \ (jsw) i}{i r_s^2 + (1-2s) X_q \ (jsw) X_q \ (jsw) i + j s r_s \ i X_q \ (jsw) + X_q \ (jsw) i} \dots (11)$$

and

$$I_{q1} = \frac{-jV_{1} I r_{s} + jX_{d}(jsw) I}{Ir_{s}^{2} + (1-2s)X_{d}(jsw) X_{q}(jsw)I + jsr_{s} IX_{d}(jsw) + X_{q}(jsw)I} + (12)$$

$$I_{d2} = \frac{V_2 I r_s + j X_q I j (2-s) v I Y_q}{r_s^2 - (3-2s) X_q I j (2-s) v I X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I j (2-s) v I + j r_s (2-s) X_q I + j r_s (2-s) X_q$$

and

$$I_{q2} = \frac{jv_2 r_s + jX_d I j (2-s) wI}{r_s^2 - (3-2s)X_d I j (2-s) wI X_d I j (2-s) wI + jr_s (2-s) X_d I j (2-s wI + jr_s (2-s)$$

XqIj(2-5)vI ...(14)

Now, the currents I_{d1} and I_{q1} alternate at a frequency sw rad/secs. while the currents I_{d2} and I_{q2} alternate at a frequency (2-s)w rad/sec Since $I_d = I_{d1} + I_{d2}$ $I_q = I_{q1} + I_{q2}$

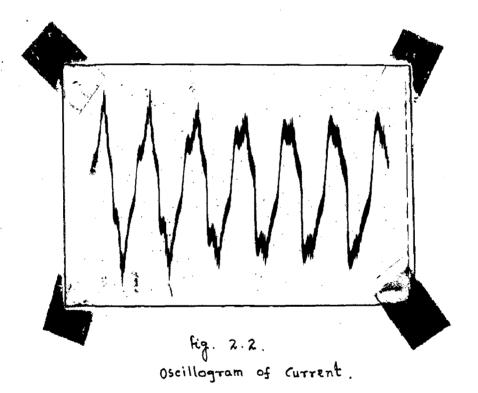
and $I_s = I_d \cos \theta - I_q \sin \theta$

Therefore, all the quantities are now again transformed to their instantaneous value.

. . 1, = $i_{d1}^{3} \cos \theta + i_{d2}^{3} \cos \theta - i_{d1}^{3} \sin \theta - i_{d2}^{3} \sin \theta$ *. $i_{d_1}^{\bullet} \cos \theta = i_{m_1} \cos (swt - \alpha_1) \cos (1-s wt)$ where α , dependents upon d-axis stator impedance. = $1_{m_1}/2$ | cos(wt- α_1)+cos (1-2s wt + α_1)] Similarly, 132005 0 =1 005 12-5 wt - 4 1005 (1-5 wt) = 1_2 loss (3-25 wt - α_3)+ cos (wt - α_3)] $-i_{al}^{s} \sin \theta = -i_{m2} \sin(svt - \alpha_2) \sin(1-svt)$ = $i_{m2}/2 \, lcos \, (wt - \alpha_2) - cos \, (\overline{1-2s} \, wt + \alpha_2)$ $-i_{g2}^{s} \sin \theta = -i_{m4} \sin(\overline{2-s} \text{ wt } - \frac{9}{4}) \sin(\overline{1-s} \text{ wt})$ = $\frac{1}{ma}/2$ [cos (wt- α_a) - cos (3-2s wt- α_a)] •• $i_s = \frac{4i_{m1}}{2} \cos(wt - \alpha_1) + \frac{i_{m2}}{2} \cos(wt - \alpha_2) + \frac{i_{m3}}{2} \cos(wt - \alpha_3) + \frac{i_{m3}}$ $1_{m4}/2 \cos(wt-\alpha_1) + 11_{m1}/2 \cos(1-2s wt + \alpha_1) - 1_{m2}/2 \cos(1-2s wt + \alpha_2)$ +11_{m3}/2 cos (3-2s wt - α_3) - 1_{m4}/2 cos (3-2s wt - α_4)1 ...(15)

Obviously the expression (15) shows that the stator current contains not only the fundamental frequency component but also the traces of the (1-2s)w and (3-2s)w frequency components.

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2.4. EXPRESSION FOR TORQUE:

In terms of the real currents and flux-linkages, the electrical torque is given by-

$$\mathbf{F}_{\mathbf{e}} = \frac{\mathbf{\Psi}}{2} \operatorname{Re} \left(\Psi_{\mathbf{d}} \mathbf{I}_{\mathbf{q}}^* - \Psi_{\mathbf{q}} \mathbf{I}_{\mathbf{d}}^* \right)$$

The mean motoring torque F_m will be equal and opposite to this torque F_e and is given by

Hence the expression for tongue with the two frequency components, can be calculated as follows:

$$V_{m} \cos swt = p \psi_{d1} + (1-s)w \psi_{q1} + r_{s} \frac{18}{d1}$$

$$V_{m} \sin swt = -(1-s)w \psi_{d1} + p \psi_{q1} + r_{s} \frac{18}{q1}$$
where $\psi_{d1} = \frac{x_{d}(p)}{w} \frac{18}{d1}$ and $\psi_{q1} = \frac{x_{d}(p)}{w} \frac{18}{q1}$

Considering the r.m.s. quantity, the vector equations are obtained by replacing p = jsw and writing capital letters for the small letters bar and putting/over flux linkage terms.

•*• $\nabla_1 = j_{SW}\overline{\psi}_{d1} + (1-s) w\overline{\psi}_{q1} + r_s I_{d1}$ $-j\nabla_1 = -(1-s)w\overline{\psi}_{d1} + j_{SW}\overline{\psi}_{q1} + r_s I_{q1}$ where, $w\overline{\psi}_{d1} = x_d \hat{v} j_{SW} I_{d1}$ $w\overline{\psi}_{c1} = x_a (j_{SW}) I_{a1}$ From (18) it can be solved for $w\bar{\psi}_{d1}$ and $w\bar{\psi}_{q1}$

From (8)-

$$\frac{v_{m}}{2} \cos(2-s)wt = p\psi_{d2} + (1-s)w\psi_{d2} + r_{s} i\frac{s}{d2}$$

$$-\frac{v_{m}}{2} \sin(2-s)wt = -(1-s)w\psi_{d2} + p\psi_{q2} + r_{s} i\frac{s}{q2}$$
...(20)

Writing in terms of r.m.s. quantity, the vector equations are-

$$v_2 = j(2+s)w \overline{\psi}_{d2} + (1-s) w \overline{\psi}_{q2} + r_s I_{d2}$$

- $jv_2 = -(1-s)w \overline{\psi}_{d2} + j(2-s)w \overline{\psi}_{q2} + r_s I_{q2}$
where $w \overline{\psi}_{d2} = x_d Ij(2-s)w I I_{d2}$

$$\mathbf{v} \overline{\psi}_{q2} = \mathbf{x}_{q} \mathbf{I} \mathbf{J} (2-s) \mathbf{v} \mathbf{I} \mathbf{I}_{q2}$$

From (20) also the values of $w\bar{\psi}_{d2}$ and $w\bar{\psi}_{d2}$ can be calculated as

$$w\overline{\psi}_{d2} = \frac{IV_2 \frac{r_8}{r_q I J (2-s) v I} + JV_2 (3-2s)I}{-(3-2s) + \frac{r_8^2}{s_q I J (2-s) w I x_q I J (2-s v I} + JV_2 (3-2s)I}$$

$$w\overline{\psi}_{q2} = \frac{JIV_2 \frac{r_8}{x_q I J (2-s) w I x_q I J (2-s) v I} + JV_2 (3-2s)I}{-(3-2s) + \frac{r_8^2}{x_q I J (2-s) w I x_q I J (2-s) v I} + JV_2 (3-2s)I}$$

...(21)

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Now, taking the conjgate of the expressions (11), (12),(13) and (14) and using the expression for torque in (17) we can get the actual expression for torque.

This expression (16) for torque may be checked by direct substitution as follows:

Let the complex quantities be-

$$I_{d} = a + jb$$

$$I_{q} = c + jd$$

$$w_{d} = c + jf$$

$$w_{c} = g + jh.$$

•*• $F_m = \frac{1}{2} (ag + bh - ce + df)$

On the other hand, the real currents and flux linkages which alternate at slip frequency are $i_{dl}^s = Re (a_1 + jb_1)e^{-jst}$

when the instantaneous torque is-

$$f_1 = (a_1 \cos st - b_1 \sin st)(g_1 \cos st - h_1 \sin st)$$
$$-(C_1 \cos st - d_1 \sin st)(e_1 \cos st - f_1 \sin st)$$

In the 2nd case when the currents and flux linkages all will vary as $e^{j(2-s)t}$, p = j(2-s)t.

•••
$$i_{d2}^s = \text{Re} (a_2 + jb_2) e^{j(2-s)t}$$

$$= a_2 \cos (2-s)t - b_2 \sin (2-s)t$$
, etc.

When the instantaneous torque in this frequency,

$$(c_2 e_2 + d_2 f_2) - (c_2 e_2 - d_2 f_2) \cos (4-2s)t + (d_2 e_2 + c_2 f_2) \sin (4-2s)t$$

.*. Instantaneous resultant torque-

$$f = (f_{1} - f_{2})$$

$$= \frac{1}{2} I(a_{1}g_{1}+b_{1}h_{1}) - (c_{1}e_{1}+d_{1}f_{1}) - (a_{2}g_{2}+b_{2}h_{2}) + (c_{2}e_{2}+d_{2}f_{2})I+I(a_{1}g_{1}-b_{1}h_{1})$$

$$(c_{1}e_{1}-d_{1}f_{1})Icos 2st+I(c_{2}e_{2}-d_{2}f_{2}) - (a_{2}g_{2}-b_{2}h_{2})cos(4-2s)t+$$

$$I(d_{1}e_{1}+e_{1}f_{1}) - b_{1}g_{1}+a_{1}h_{1})Isin 2st+I(b_{2}g_{2}+a_{2}h_{2}) - (d_{2}e_{2}+c_{2}f_{2})$$

$$-t-(4-2s)t I$$
(22)

Thus an expression for torque has been developed directly in terms of the components of the compex number form of the d-axis and q-axis quantities. But it is very difficult to get the exact expression for the torque. However, from the above expression (22) one can get an idea about the torque-slip characteristic %of the motor.

<u>CHAPTER-III</u>

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3.1. EQUIVALENT CIRCUIT:

All the rotating machines behave in accord with immutable laws of physics. In them electrical, magnetic, and mechanical energies co-exist and interact. Two methods have been followed to analyze these electromagnetic and mechanical systems. One method is to represent the system by differential equations, carried to the required degree of exactness and to get the numerical solution of these equations. The second method is to represent the system or its differential equations by an equivalent circuit, consisting of stationary electric-circuit elements, which can be calculated theoretically and verified experimentally.

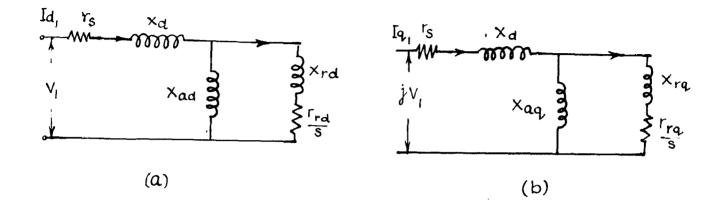
In order to calculate the torque-speed curve of the machine, it is easier to use the equivalent circuits instead of the expression (17)(Chapter II, page)), The method, similar to that commonly used for induction motor, has been applied to the two fletitious axes circuits, namely d-axis and q-axis circuits. Following the operational equa.(7) and (3) (Chapter II, page)) and reducing them to the r.m.s. quantities of A.C. systems, the following equivalent circuits as shown in fig.3.1 can be developed. Thus each of the component torques can be determined from the appropriate equivalent direcuit.

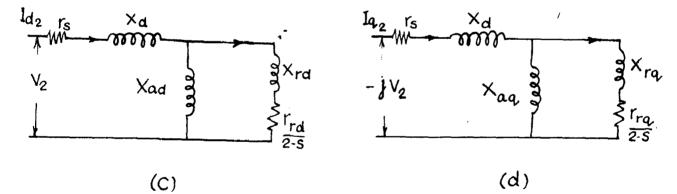
From the equivalent circuits of Fig.(3.1) the values of I_{d1} , I_{a1} , I_{d2} and I_{g2} can be formed out as follows:

$$I_{d1} = \frac{V_{1}}{(r_{0} + 3E_{1}) + \frac{3 I_{cd} (P_{rd} + 3 E_{rd})}{P_{pd} + 3 (I_{cd} + \frac{1}{2E_{d}})}}$$

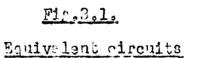
$$I_{q1} = \frac{3V_{1}}{(r_{0} + 3E_{1}) + \frac{3I_{cd} (P_{rd} + 3 E_{rd})}{P_{pd} + 3 (I_{cd} + \frac{1}{2E_{d}})}}$$

$$= \frac{3V_{1}}{P_{pd} + 3 (I_{cd} + \frac{1}{2E_{d}})}$$





(C)



$$I_{d2} = \frac{V_2}{(r_s + jx_1) + \frac{jx_{ad} (r_{rd} + j \frac{x_{rd}}{(2 - s)})}{r_{rd} + j(x_{ad} + \frac{x_{rd}}{2 - s})}}$$

$$I_{q2} = \frac{-jV_{2}}{(r_{s}+jX_{1})+\frac{jx_{aq}(r_{rq}+j-\frac{x_{rq}}{2-s})}{r_{rq}+J(x_{aq}+\frac{x_{rq}}{2-s})}}$$

Using these values of I_{d1} , I_{q1} , I_{d2} and I_{q2} the expression for the torque can be developed.

$$P_{d1} = I_{d1}^{2} \frac{r_{rd}}{s}$$

$$F_{d2} = I_{d2}^{2} \frac{r_{rd}}{2-s}$$

$$F_{q1} = I_{q1}^{2} \frac{r_{rd}}{s}$$

$$F_{q2} = I_{q2}^{2} \frac{r_{rq}}{2-s}$$

and

 $F_q = F_{q1} + F_{q2}$

In an ordinary induction motor the air-gap is uniform, and hence the magnetic circuit is symmetric about any axis. But due to saliency the magnetic circuit will have two axes of symmetry. The presence of saliency in the magnetic circuit of an electrical machine affects the reluctance of the magnetic circuit by the so called 'fringing of flux' effects. The flux lines are ratified

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in the interpolar region. The reluctance of the air-path is increased and thus some flux is lost. This effect can be considered as due to the slot-opening in the turbb generator rotor. The increased reluctance is attributed to an increase in the effective air-gap under polar and inter-polar region. A method with the help of Fourier series representation of air-gap flux is outlined, from which gap co-efficients in the two axes are obtained.

Now, putting the values from Fig.(A_1), we can calculate, $C_A = 1.04$

and C₀ = 1.90

So long we have considered only the saliency of the rotor, but the slotting in stator will also attribute something in 'flux fringing phenomenon. This can be taken into account as a further elongation of the effective gap-length in both polar and inter-polar region. This factor can be calculated by formula in Say's book.

$$K_g = \frac{y_g}{y_g - K v}$$

R_R is calculated as 1.10.

Calculation of X_{ad} and M_{mq} #

The formula given by Veinott to calculate the magnetising reactance has been used. Only the air-gap length in the proper axis has been taken into consideration by multiplying the gap-length by the corresponding gap co-efficient.

Formula given by Veinott,

 $X_{\rm H} = 2 \ \pi \ f \ (CK_{\rm H})^2 \ \pi \ \frac{O_{\rm e} 411}{p} \ \pi \ \frac{\Lambda_{\rm R}}{B_0} \ \pi \ 10^{-8} \ \text{ohms.}$ Hence this has been modified as- $X_{\rm md} = 2 \ \pi \ f \ (CK_{\rm H})^2 \ \pi \ \frac{O_{\rm e} 411}{p} \ \pi \ \frac{\Lambda_{\rm R}}{R_{\rm e}} \ \pi \ 10^{-8} \ \text{ohms.}$ and $X_{Eq} = 2 \ f(CK_{y})^2 \equiv \frac{0.411}{p} \equiv \frac{A_R}{C_0 \pi c_q} \equiv 10^{-8} \text{ ohms.}$ where, $CK_y = \text{Effective series conductors.}$ f = frequency in cyclos/sec. p = no. of poles. $A_g = \text{Air-gap area (in)}^2$ $G_0 = \text{Equivalent air-gap length (in).}$

Putting the values of the proper constants we have calculated~

X_{md} = 550 ohns X_{mg} = 294 ohns.

Stator and Rotor Leakage reactancess

The stator leakage is calculated following the approach of Veinott⁸. The total leakage of the stator winding is taken as the sum of stator slot leakage, half the sigsag, belt and the end leakage. The skew leakage is neglected since the machine fabricated is not having the skew. The leakage reactance of the &-axis and q-axis robor windings are calculated following the approach of Talhat.⁹ Only the formulae in their final form are presented here. For detail ed derivation the original papers may be considered. As far as possible, symbols used by the above authors have been used. To keep the sequence a few symbols used by Talaat has been changed. Both the above authors used inch as the unit of dimension and the same has been retained here.

Stator Leakages

Stator leakage reactance $X_1 = K_{\pi} \pi$ (stator leakage constant) where $K_{\pi} = 2 \pi f(CK_{\mu})^2 \pi 10^{-8}$

Stator loakago constant = stator slot of (Bolt+slgzog +End).

Stator slot = $\frac{3.19 \text{ m K}_{s1} C_x L_1}{s_1}$

where, m = no. of phases to be taken as 2 for single-phase motor. $K_{s1} \equiv \text{Stator slot permeance (see fig.17.10,p.329 Weinott).}$ $C_{x} = \text{Corrector factor.}$ $L_{1} = \text{Stator stack length (in).}$ $s_{1} = \text{No.of stator slot.}$ (Belt' = 0.00119 m.K_mK_B where $K_{m} = \frac{A_{x}}{E_{c} \otimes F_{m} p}$. $SF_{m} = \text{Saturation factor}$ $= \frac{\text{Air-gap AT + Iron AT}}{\text{Air-gap AT}}$

K_B is belt leakage constant to be read from Fig.17.12 of Veinott's book.

"Zigzag" =
$$\frac{1.065 \text{ mL}_1}{\text{S}_1\text{S}_6}$$
 K_{2s}
when K_{zz} = $\frac{(t_{10} + t_{20})^2}{4(\lambda_1 + \lambda_2)}$
where, t_{10} = stator tooth face (in)
 t_{20} = Rotor tooth face (in)
 λ_1 = Stator tooth pitch (in)
 λ_2 = Rotor tooth pitch (in)
 λ_2 = Rotor tooth pitch (in)
 $\frac{1.57m D_0 (ACT)}{s_1 p}$

where $D_c = Diameter$ at the centre of stator slots ACT = Average Coil throw.

for the experimental machine

K_x = 2.512 Stator slot=1.27 'Belt' = 1.25 'Zigzag' = 4.16 'End' = 2.82

.*. Stator leakage constant = 1.27 + 1(1.25 + 4.16 + 2.82)

#1.27 + 4.11 = 5.38

.*. Stator leakage reactance

 $X_L = K_x x$ (Stator leakage constant)

= 2.512 x 5.38 = 13.5 ohms.

Direct-Axis Rotor winding Leakage Reactance:

The formula given by Talaat⁹for this reactance is re-arranged as follows:

$$X_{2d} = 2 \pi f(CK_y)^2 x \frac{2}{p} x \frac{L_2 x}{h_b} \frac{1}{(1-k_b)} x 10^{-8}$$

where,

 $n_b = No.$ of bars per pole.

"Tooth Top! = 3.19 x P_{ub} when R_{tb} is to be read from Fig.2 of Talaat's paper.

(Neglect this for closed rotor slots) 'Harmonics' = 3.19 x $\frac{Tb}{12.8c}$

where, $T_b = \text{Rotor slot pitch (in)}$ 'End' = 3.19 x $\frac{0.12}{m}$ x $\frac{T_r}{T_b}$ x $\frac{T_{ring}}{L_2}(2 + \frac{\cos n_b \alpha_b - K_b \cos \alpha_b)}{1-K_b}$

where T_{μ} = Pole pitch at rotor surface (in)

Tring[#] Pole pitch at average ring diameter (in) Other symbols have been defined earlier.

For the experimental machine, these values of constants are calculated as follows:

'Rotor slot'	-	31.9	hb	8	15
'Harmonics'	3	4.6	K,	Ħ	0.205
*End *	*	3.4			

••• $\Lambda_{bed} = 31.9 + 4.5 + 3.4 = 39.8$ ••• $X_{Fd} = 2\pi f = x L_2 \times (CK_y)^2 \times \frac{2}{p} \times \frac{\Lambda_{bed}}{n_b(1-K_b)} \times 10^{-8}$ ohms.

= 25 ohms.

Quadrature Axis Rotor Winding Leakage Reactance:

The formula for this reactance as given by Talaat is as follows:

$$X_{rq} = 2 \pi f(CK_{v})^{2} x L_{2} x \frac{2}{p} x \frac{\hbar bed}{n_{b}(1+K_{b})} x 10^{-8} ohns.$$

where A_{beq} is the permeance constant for the quadrature axis winding. Other symbols have been already defined earlier.

Like \bigwedge_{bed} , \bigwedge_{bed} consists of four components.

> *Botor slot* = 41.5 $n_b = 15$ *Harmonics* = 4.5 $K_b = 0.205$ *End* = 3.4 .*. $k_{beq} = 41.5 + 4.5 + 3.4 = 49.4$.*. $X_{20} = 20.4$ ohms.

Stator and Rotor Resistance Calculation:

Stator Resistance r_{g} : The stator resistance can be calculated by knowing the length of the mean conductor. This will depend to a certain extent, on the method of manufacture.

The formula given by Veinott is as follows: Length of the mean conductor = LMC = $L_1 + \frac{\sqrt[3]{(ACD)} - D_2}{S_1}$

where 3 = Emperical constant

= 1.30 for two pole design

= 1.50 for four pole design

= 1.70 for six or more pole design.

Other symbols have already been defined earlier for the experimental machine.

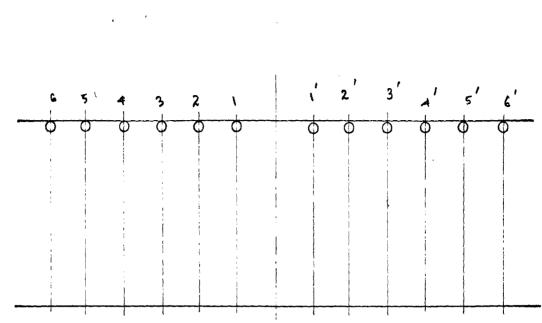
LMC = 10.32 in

.". The total length of the winding wire

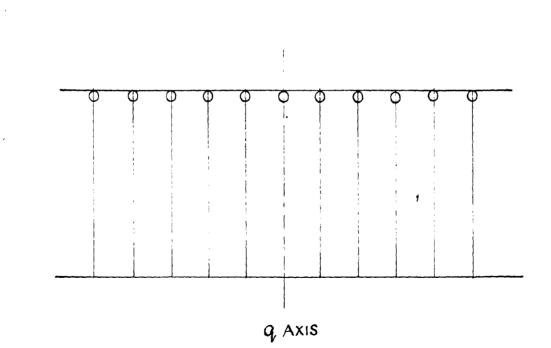
= No. of conductors x LMC

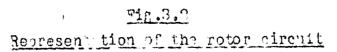
= 296 yards.

The wire size used is 22 S.W.G. and from the table given by the manufacturer, the resistance for this length at 25'C is 12.10hms.









.*. r_g = 1211 ohms.

Direct Axis Rotor Winding Resistance, r24:

The formula given by Talaat is as follows:

 $r_{2d} = (CK_y)^2 x - \frac{2}{p} - \frac{r_{bed}}{n_b (1-K_b)}$

 r_{bed} is resistance constant to be evaluated. Talaat has not given any formula for r_{bed} . It would be equal to the resistance of a bar if the end-ring resistance can be neglected. This may be case in a synchronous machine with a large no. of poles. But for the fractional motors with very small no. of poles, the endring resistance can not be neglected. In deliving at the expression for r_{bed} , the following assumptions as made by Talaat has been retained.

- 1) The space distribution of bar currents are simusoidal.
- 2) A sinusoidal pulsating flux density is assumed to be entered on the direct-axis.

According to the assumptions made, the current in bars θ_1 (1-1') would be $i_{Dd} \sin \frac{\theta_1}{2}$ where θ_1 is the electrical space angle between bars (1-1') and I_{Dd} is the current that would be flowing in a bar situated at 90° (electrical) from the d-axis.

Similarly, the currents flowing in bars (2-2*), (3-3*), would be $I_{Dd} \sin \frac{\theta_2}{2}$, $I_{Dd} \sin \frac{\theta_3}{2}$ etc.

If r_b is the resistance of each bar, the Hosses taking place in bars in one pole pitch would be given by the following.

Losses in bars (per pole pitch)

= 2 r_b ($I_{Dd}^2 \sin^2 \frac{\Phi_1}{2} + I_{Dd}^2 \sin^2 \frac{\Phi_2}{2} + ...$) The terms in the bracket have been summed up by Talaat as The losses in the end-rings can be calculated as follows: The current flowing in sector n and (n-1) of the end-ring is $I_{Dd} \sin \frac{0}{2}$). The total losses in the ring can be found out by finding out losses in the individual sections.

Let the total losses in end-rings be denoted by Pring,

The total losses in the rotor are $(P_{bas} + P_{ring})$. Equating the total losses to expression (79) of Talaat's paper, which gives the total losses in terms of r_{bed} we get,

$$P_{bar} + P_{ring} = \oint r_{bed} I_{Dd}^2 n_b (1 - K_b)p_*$$

from this, substituting the values of Pber and Pring,

$$r_{bed} = r_b + \frac{r_{end}(d)}{\frac{1}{2} n_b (1 - K_b) p}$$

For the experimental machine the values has been calculated

es $r_b = 1.23 \times 10^{-4}$ ohms $r_{sector} = 5.21 \times 10^{-6}$ ohms. $r_{end}(d) = 2.51 \times 10^{-4}$ ohms. .*. $r_{bed} = 1.45 \times 10^{-4}$ ohms. and $r_{rd} = 10.5$ ohms.

Quedrature Axis Rotor Winding Resistance:

The formula given by Talaat for this resistance is-

 $r_{gq} = (CK_y)^2 \times \frac{2}{p} \times \frac{r_{beq}}{n_b} (1+K_b)$

Talaat has not given expression for r_{boq^2} but it can be found in exactly the same fashion as the expression for r_{bed^2}

$$e^{\circ} P_{beq} = P_b \diamond \frac{P_{ond}(q)}{\& P_b} (1 \diamond R_b)p$$

 $r_{end}(q)$ is found in the same manner as $r_{end}(d)$.

For the experimental machine the following values have been computeds

$$r_b = 1.23 \text{ m } 10^{-4} \text{ ohmo}$$

 $r_{\text{sector}} = 6.11 \text{ m } 10^{-6} \text{ ohms}$
 $\circ \circ r_{\text{end}}(q) = 1.10 \text{ m } 10^{-4} \text{ ohms}$
 $\circ \circ r_{\text{boq}} = 1.83 \text{ m } 10^{-4} \text{ ohms}$
 $\circ \circ r_{\text{boq}} = 1.83 \text{ m } 10^{-4} \text{ ohms}$

Moment of Inertia:

The moment of inertia of the shaft and the rotor can be easily calculated following the well-known formula for Hollow cylinder.

Moment of inertia for hollow cylinder $rac{M(R^2 \diamond r^2)}{2}$

where M = Mass of the cylinder

R = External radius

r = Internal redius

For the solid cylindor 'r' would be zero. The moment of inertia of the shaft of the experimental machine was calculated as 0.166 in²-1b. After making suitable corrections for the presence of saliency and slots, end rings etc. the moment of inertia of the rotor is found to be 11.743 in^2 - 1b. The total inertia for the shaft and the rotor is thus found to be 11.743 in^2 .lb. which is equivalent to $35 \times 10^{-4} \text{ Kg} \cdot \text{m}^2$.

<u>CHAPYER-IV</u>

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G.L. STUDY OF ELECTRO MECHANICAL TRANSIENT:

The application of induction motor in control systems has lod to the study of dynamic behaviour of induction motor in which non-linear equations must be solved. It involves the separate solution of the equations for all the inputs of interest. Digital computer is now-a-days gaining popularity, because it does all the mathematical manipulation of numerical solution in a shorter time. But analog computer is also a powerful tool, which helps simulating the actual system with electrical or electromechanical components, Other methods of investigating the dynamic behaviours are phase-plane tochniques, describing functions, quasi-linearizations and statistical mothods. These methods are mainly concerned with the consideration of stability in non-linear systems.

4.2. ANALOGUE COMPUTER:

When a machine is viewed from the standpoint of dynamic circuit theory, a set of integro-differential equations of greater or less complexity depending on the machine predict the whole transient performance of the machine. The solution of these non-linear equations are frequently a formádable task. A very convenient method for studying simulataneous ordinary intregro-differential equations (both linear and non-linear) is the electronic analogue computer. The analogue computer consists of components of various types that can be interconnected to simulate the operation of the system in time domain.

Considering the Fig. (2.1) and with the help of dynamic circuit theory the following equations can be developed.

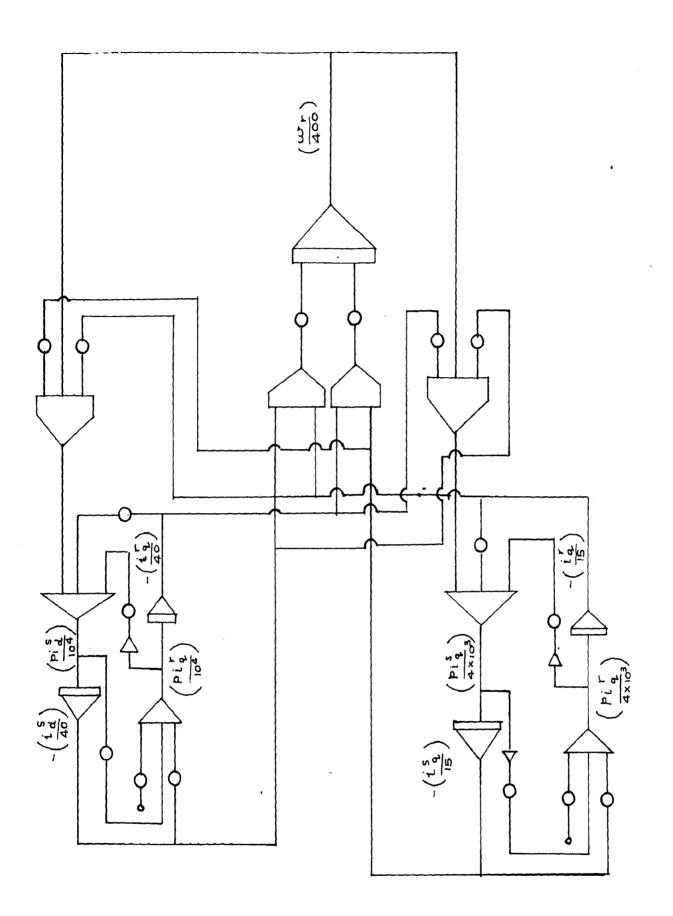
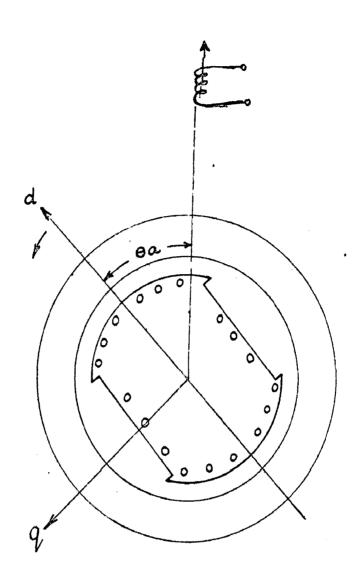


Fig.4.1

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Representation of Experimental wiching

$$r_{s}i_{d}^{s} + L_{d} \frac{di_{d}^{s}}{dt} + L_{ad} \frac{di_{d}^{r}}{dt} = v_{d}(t)$$

$$L_{addt}^{d} + L_{rd} \frac{di_{d}^{r}}{dt} + r_{rd}i_{d}^{r} + v_{r} (L_{rq}i_{q}^{r} + L_{aq}i_{q}^{s}) = 0$$

$$r_{s}i_{q}^{s} + L_{q} \frac{di_{q}^{s}}{dt} + L_{aq} \frac{di_{q}^{r}}{dt} = v_{q}(t)$$

$$L_{aqdt}^{d} + L_{q} \frac{di_{q}^{r}}{dt} + L_{aq} \frac{di_{q}^{r}}{dt} = v_{q}(t)$$

$$L_{aqdt}^{d} + L_{rq} \frac{di_{q}^{r}}{dt} + r_{rq} i_{q}^{r} + v_{r} (L_{rd}i_{d}^{r} + L_{ad} f_{q}^{s}) = 0 \qquad \dots (1)$$

$$T_{D} = (i_{d}^{s}i_{q}^{r} L_{aq} - i_{q}^{s}i_{q}^{r} L_{ad})$$

$$= J_{e} \frac{dv_{r}}{dt} + T_{L}$$
if $T_{L} = 0$ and $\theta = 45^{\circ}$

The analogue set-up given in Fig. (4.2) can be developed.

4.3. DIGITAL COMPUTER STUDY:

Following the Fig.(4.1), the voltage equations can be written in terms of flux-linkage also.

$$r_{s}i_{s} + \frac{d\psi_{s}}{dt} = V_{m} \sin (wt + \beta)$$

$$r_{rd}i_{d}^{r} + \frac{d\psi_{r}d}{dt} = 0$$

$$r_{rq}i_{q}^{r} + \frac{d\psi_{r}q}{dt} = 0$$

$$T_{D} = J \frac{d^{2}\theta}{dt^{2}} + T_{L}$$

The flux-linkages can be written as

$$\Psi_{s} = L_{s1_{s}} + L_{ad} \cos \theta \, \mathbf{i}_{d}^{r} - L_{aq} \sin \theta \, \mathbf{i}_{q}^{r}$$
$$\Psi_{rd} = L_{ad} \cos \theta \, \mathbf{i}_{s} + L_{rd} \, \mathbf{i}_{rd}$$

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...(2)

$$\Psi_{rq} = -L_{aq} \sin \theta i_s + L_{rq} i_q^r$$

Now, the electro-dynamic developed torque-

$$T_D = \frac{V_{Ploid}}{9}$$

and the energy stored in the electromagnetic field-

••• $T_D = -L_{ad}$ is i_d^T sin $\theta - L_{aq}$ is $i_q^T \cos \theta$ Now, substituting the values for $\frac{d\psi_i}{dt}$, $\frac{d\psi_{rd}}{dt}$ and $\frac{d\psi_{rq}}{dt}$ and writing $\frac{d\theta}{dt} = w_r \& \frac{d}{dt} = p$ the expression (2) can be constrained to a set of first-order non-linear differential equations in the form.

L _g	Lad cos 0	- L _{ag} sin 0	pis
Ladcos 9	^L rđ	0	pia
-L _{aq} sin 9	0	Lrq	piq

$$Vsin(vt+\emptyset) = r_s i_s + L_{ad} i_d^T w_r sin \Theta + L_{aq} i_q^T w_r cos \Theta$$

$$-r_r d i_d^T + L_{ad} i_s w_r sin \Theta$$

$$-r_r q i_q^T + L_{aq} i_s w_r cos \Theta$$

$$r_r q i_q^T + L_{aq} i_s w_r cos \Theta$$

$$r = \frac{1}{2} (T_L + L_{ad} i_s i_d^T sin \Theta + L_{ad} i_s i_d^T cos \Theta)$$

$$pv_{r} = \oint (T_{L} + L_{ad} i_{s} i_{d} sin \Theta + L_{ad} i_{s} i_{d}$$
$$p\Theta = v_{r}$$

These non-linear differential equations are solved in digital computer by Runge-Kutta-Gill method. The computer-program has been incorporated in Appendix III. 105101 3 100 concerns incorporated in Appendix III. 105101 3 100 concerns incorporated in Appendix III. 105101 3 100 concerns incorporated in Appendix III. 105101 3 100 concerns

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. <u>CHAPTER - V</u>

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5.1. EXPERIMENTAL DETERMINATION OF EQUIVALENT CIRCUIT CONSTANTS:

The method of experimentally determining the machine constants of 3-phase induction motor is quite well known. In case of single phase induction motors, the problem of determining the motor constants is not so simple. It may be noted that for a single phase motor under no-load conditions, the rotor slip with respect to the forward field is nearly zero, but is nearly double with respect to the backward field. Hence there are substantial rotor cubploss even at no load. It may be also noted that in the usual revolving field equivalent circuit, helf of the magnetising reactance is nearly short circuited at the time of no-load test. Veinott has given a test method which takes care of the above points and his books may be referred for the details of this method.

The separation of stator and rotor leakage reactance for squirrel cage induction motor is quite a difficult problem. No simple method is available for this separation. Hence the comparison between the theory and the experiment is also very difficult. It is well known that the running performance calculation is of less importance for the normal motor. The starting performance is the prime important factor. In our machine also starting performance is very interesting. The starting torque arises due to different impedances of the two rotor circuits. The difference between winding impedances must be known as accurately as possible.

For the present machine, it is considered desirable that some method may be approximate one, shall be used to determine the leakage separately. The stator leakage reactance is measured separately by a test called the 'rotor out' test. Besides this test, the 'synchronous reactance' of the machine was measured by 'maximum reluctance' test. Magnetising reactances for both the axis was determined by substracting leakage reactance from the 'synchronous reactances'.

The rotor impedance is calculated from the locked rotor test. The rotor impedance is different in the two axis. Hence the lockedrotor test is done twice, once with the d-axis of the rotor aligned with the stator winding axis and the other is with q-axis of the rotor aligned with the stator winding operational impedance is calculated from frequency response method. The moment of inertia is measured by 'decceleration test'.

'Rotor Out' Test:

In this method, the rotor with shaft is removed from the motor. End shields are kept fitted. The reactance of the stator winding is measured under this condition. This reactance is the sum of the stator leakage and the magnetising reactance. The magnetising reactance is due to the small amount of mutual flux going through the very large air-gap resulting from the removal of the rotor. It is nocessary to find the reactance due to this flux. This can be calculated from Veinott's formula for the magnetising reactance with the modification as 'g'= the air-gap length as the ratio of the stator bore and no, of poles. The modified formula becomes-

 X_{m} (rotor out) = 2 π f (CK₀)² π 0.6468 $\pi - \frac{L}{P} = \pi 10^{-8}$ ohms.

where, f = frequency in c/s.

L = Length of the stator in inch.

CK, = Effoctive conductors in series.

P = no.of poles.

The resistance and reactance are calculated from the readings taken. The value of X_{\Box} (rotor out) is subtracted from the measured total reactance to get the leakage reactance x_1 .

Test Readings:

Voltage	*	37 volto.
Current		1.5 Ampr
Power		30 water.
X Total	#	20.8 ohun.
X _{m(rotor}	ont) *	5.05 ohm.

**. X1 = Xtotal - Xn (rotor out) = 15.75 chun.

The d.c. stator resistance is found. The a.c. resistance measured in this test includes a small iron loss component. After allowing for this loss also the resistance is found to be greater than d.c. value which can be accounted for the skin effect.

Maximum Reluctance Test:

The present machine when driven at synchronous speed represents the reluctance motor. The load is gradually increased and at the break point from synchronous to induction run, the power input, the current and the voltage are noted.

For maximum power condition-

$$P_{\rm m} = \frac{{\bf v}^2}{2} \left(\frac{1}{{\bf x}_{\rm q}} - \frac{1}{{\bf x}_{\rm q}} \right)$$

and
$$I = \sqrt{I_d^2 + I_q^2} = \frac{V}{\sqrt{2}} \sqrt{\frac{1}{I_q^2} + \frac{1}{X_q^2}}$$

Substituting $X_{d}/X_{q} = K$, there are,

$$P_{\rm m} = \left(\frac{\sqrt{2}}{2X_{\rm q}}\right) \left(1 - \frac{1}{K}\right) \qquad \dots \qquad \dots (1)$$

I = $\left(\sqrt{\sqrt{2} X_{\rm q}} K\right) \left(\sqrt{1 + K^2}\right) \qquad \dots \qquad \dots (2)$

From (1) & (2),

K-] = /2 P./VI = 8

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$$K = \frac{1 \pm a \sqrt{2-a^2}}{(1-a^2)}$$

That value of K which is greater than one is retained and X_q is calculated from either (1) or (2).

Then X_d is calculated, as, X_d = KX_g Test Results:

Pover input in vatts	=	33
Voltage in volts		295
Current in Amperes	ø	0.70
R is calculated in	6	1.39

and X, 10 calculated as 319 ohms.

. Xa = KXa = S14 ohms

Lockod Rotor Test:

The locked rotor tests are done twice once with d-axis of rotor aligned with stator axis and the other with q-axis of rotor aligned with stator axis. Keeping the rotor in locked position, in us annually the rotor is turnod gradually to get the maximum deflection showing the alignment of q-axis with stator axis and minimum for d-axis respectively.

The test results for the experimental machine are given below: A) Test results with minimum deflection

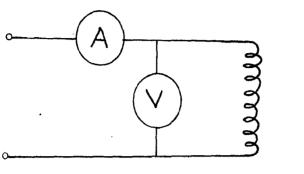
Power in watto=62Voltage in volta=85Current in Amps.=1.5

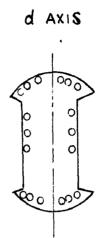
B) Test results with maximum deflection.

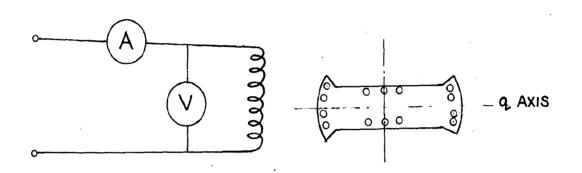
Power in vatto D 132

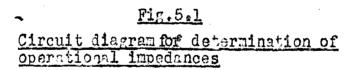
Voltago in voltan 96

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From the locked rotor tests (rated current) the values of r_{rd} , π_{rd} , r_{rq} , π_{rq} , are obtained. These are obtained by calculating the total locked rotor impedances and subtracting the stator impedance value from them. The effect of magnetising reactance is neglected.

From the d-axis test,

$$Z_{\mathbf{q}} = \frac{W}{\mathbf{1}^2}$$

$$R_{\mathbf{q}} = \frac{W}{\mathbf{1}^2}$$

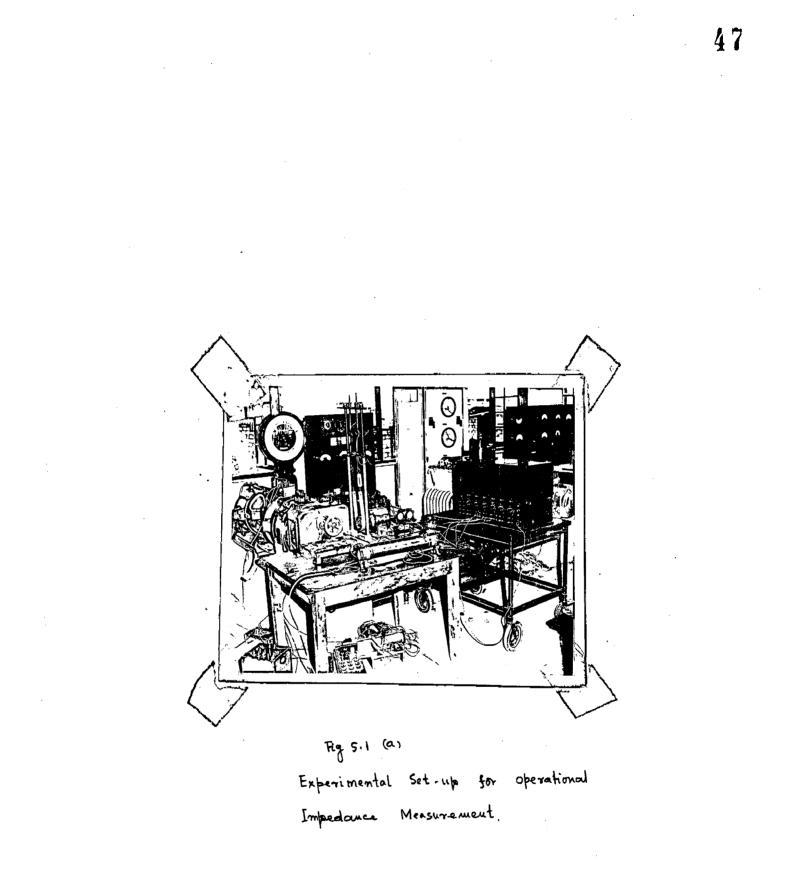
$$R_{\mathbf{q}} = \sqrt{Z_{\mathbf{q}}^2 - R_{\mathbf{q}}^2}$$

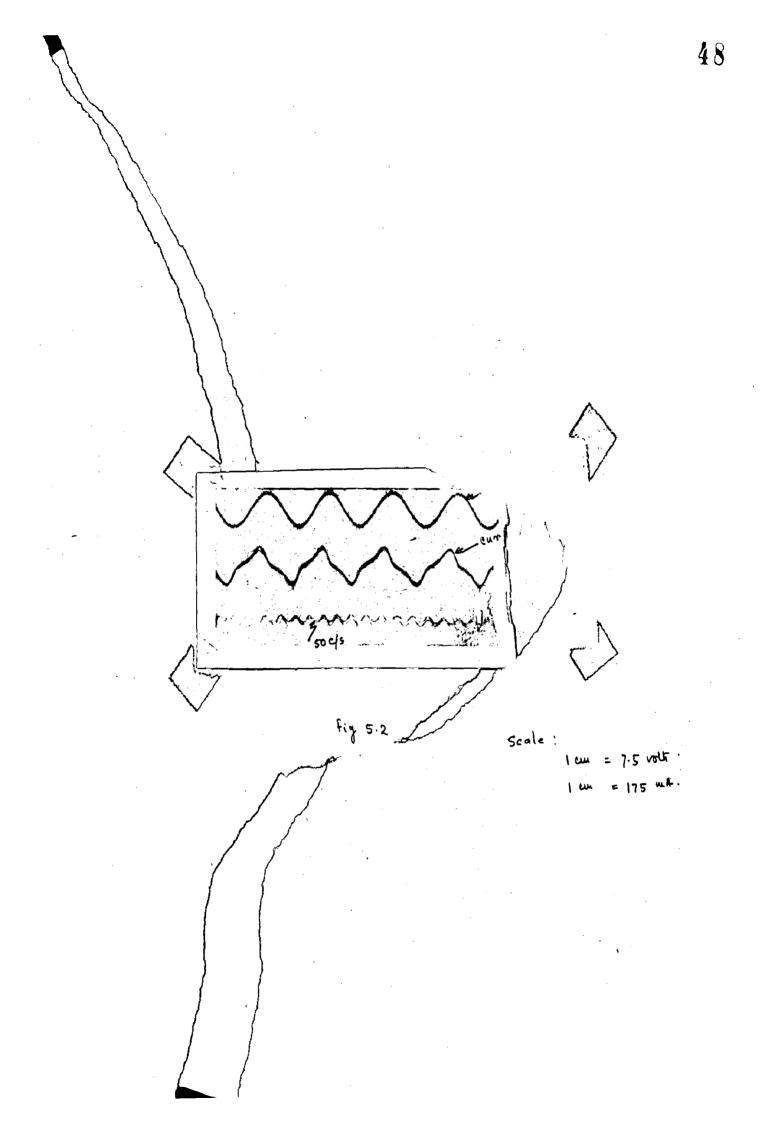
The stator impedance from the "rotor-out" test

 $r_{g} * jr_{g}$ $r_{g} * jr_{g}$ $r_{rd} = R_{d} = r_{g}$ $r_{rg} = 10.6 \text{ ohm}$ $r_{rd} = 14.3 \text{ ohm}$ $r_{rg} = 10.6 \text{ ohm}$ $r_{rd} = 33.5 \text{ ohm}$ $r_{rg} = 18.5 \text{ ohm}$

8.2. DETERMINATION OF OPERATIONAL IMPEDANCES:

The torque expression (17) of Chapter I has been developed in term of operational impedances. So the calculation of torque-slip characteristic domands the determination of operational impedances. The method presented here is a slight modification of that already presented by Son & Adkins. If the input impedance at a frequency sv rad/occ. with rotor standstill is Z_{0} for direct axis in line with the stater m.m.f. axis (position being determined by indication of





minimum dofloction in the assotor), then,

$$Z_{0} = F_{0} + j_{0} = \frac{(j_{0}v)}{X_{0}} = \frac{1 + j_{0}v}{1 + j_{0}v}$$
where $X_{0}, (j_{0}v) = \frac{1}{X_{0}} \frac{1 + j_{0}v}{1 + j_{0}v}$

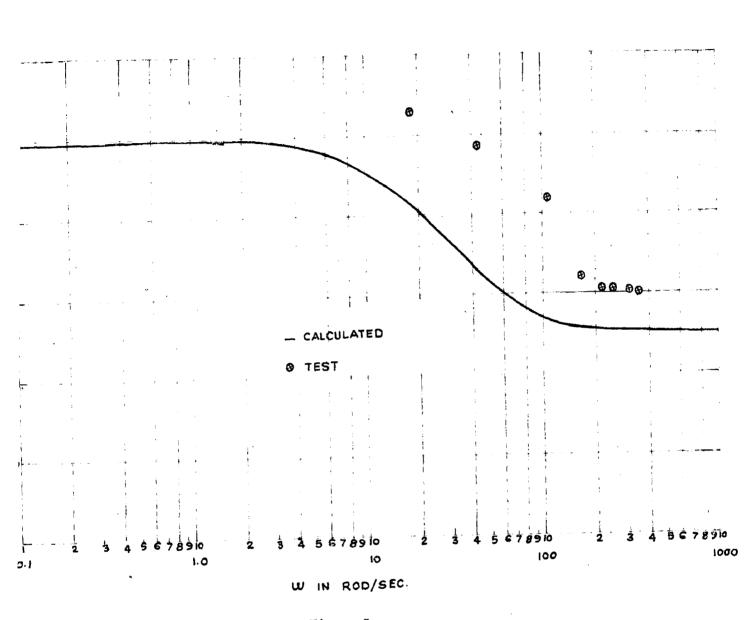
Similarly, the operational impodance X_{gl}(jow) is obtained from the following,

Z_q = r_s + jo X_q (jov)

where Z_q is the input impedance at a frequency sw rad/sec. with rotor standstill for quadrature axis in line with the m.m.f. axis (position being determined by the maximum deflection of ammetor)_p

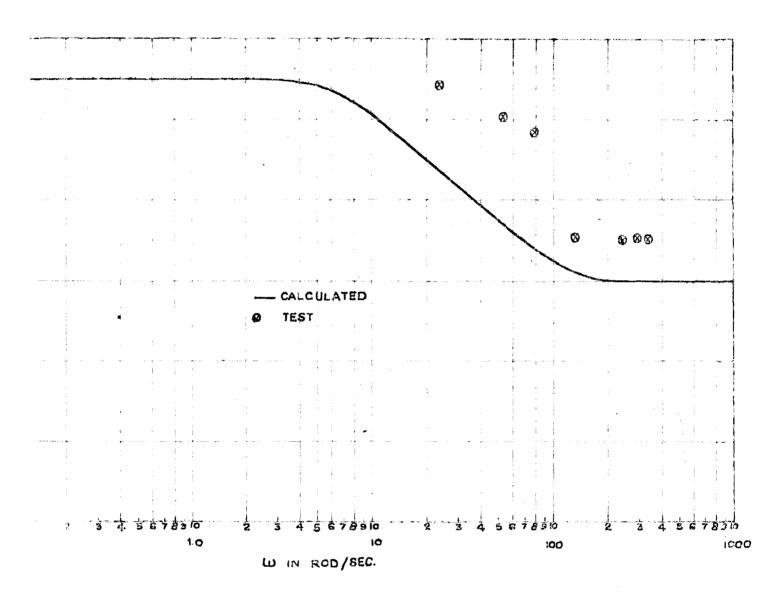
$$X_q$$
 (jsu) = $X_q = \frac{1 \diamond jsu T_q}{1 \diamond jsu T_{q0}}$

The connections are shown in Fig. (5.1). The frequency is varied from very low value to rated value, and the input impedances are measured. The magnitude and phase difference between the voltage and the current is measured with the help of a six element Cambridge Duddel Oscillograph. Variable frequency supply is obtained from a separately driven three phase commutator machines. A typical oscillogram of current and voltage is given in Fig. (5.2). The amplitudes of $X_{\rm G}$ (jew) and $X_{\rm G}$ (jew) are plotted against $\log_{10}(4w)$ Fig. (5.3) & (6.4). The inverse of corner frequencies obtained from their straight line approximation will yield the four time constants $T_{\rm do}^{i}$, $T_{\rm do}^{$



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Ploy of wolling x (p) in terms of log-frequency



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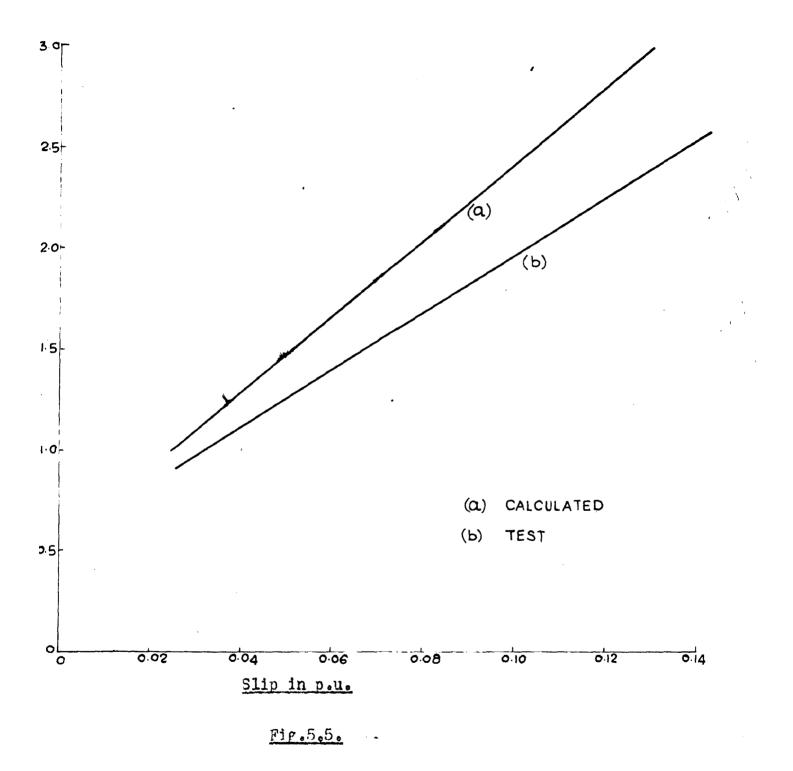
5.3. DETERMINATION OF MOMENT OF INERTIA BY DECCELERATION TEST:

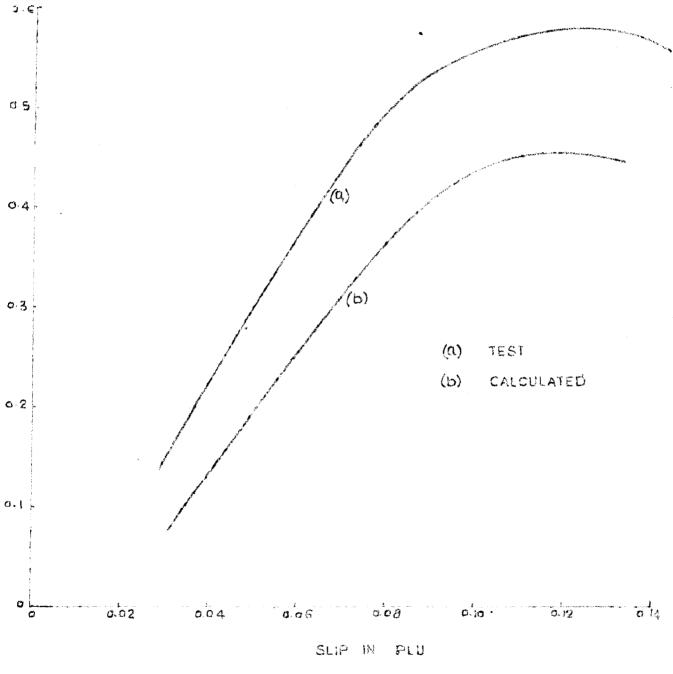
The moment of inortin has been measured by the deceleration test. The rotor is allowed to slow down from no-load to lower-speeds due to friction. Knowing the friction torque, the moment of inertia can be found out. This can not be very accurate but has been used as a check on the theoretically calculated value. The theoretical value can be calculated accurately for the rotor and shaft.

Experimentally the value has been found out as $3.1 \times 10^{-3} \text{ Kg} \cdot \text{m}^2$.

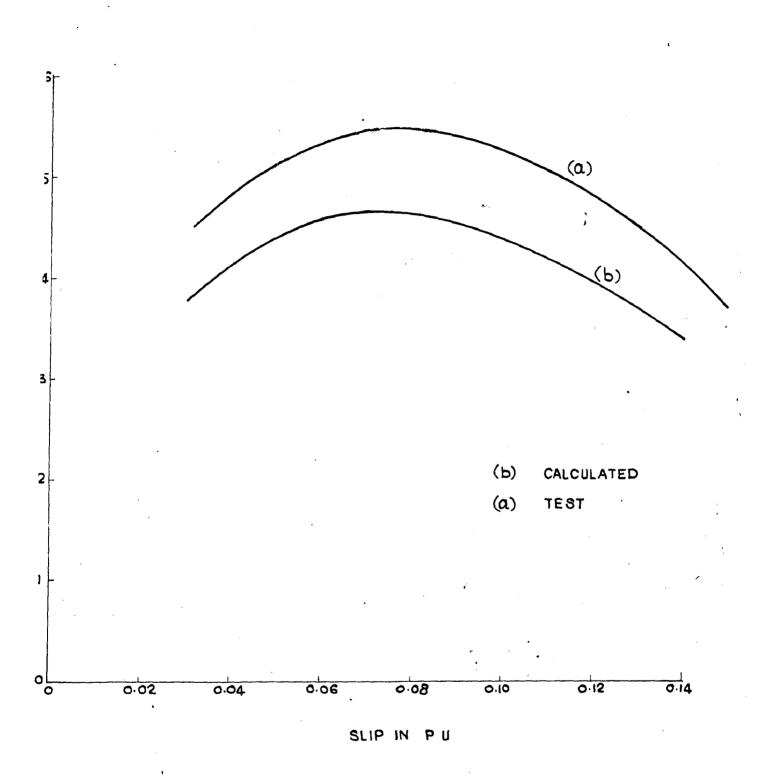
5.4. LOAD TEST:

The dynamometer required for the test machine is not available. Hence the load test has been carried out by the 'pulley-belt' method. The belt-ends are connected to the two spring balances and the motor is loaded by the usual method. The test results are shown in a tabular form below. From the results, curves shown in figs. (5.5), (5.6), A (5.7) are drawn.





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V in Volts; I	and da sa su anna an da ba an an a' a dha an dha an dha		Pulley	Pulley diameter - 3+"		
	in Amps in	watts 1	N . a r.p.m. in	T ₁ , lbs.	T2 in 1bs.	
a)	220	1.0	108	2920	0	0
b)	250	1.1	136	2910	0	1
c)	220	1.25	180	2890	0	1.5
đ)	250	1.6	220	2850	0	8
e)	220	1,85	316	2790	0	3
f)	550	2,1	364	2750	Ó	3.5
g)	220	3.0	520	2610	0	4.0
				, ·		

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TEST RESULTS OF THE EXPERIMENTAL MACHINE:

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CONCLUSION

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MAIN PEATURES OF THE MOTOR DEVELOPEDA

The principlo of using asymmetry of the rotor for starting a single-phase induction motor is developed and has been found to be successful.

The method of construction is casier and eliminates the use of auxiliary vindings and shaded rings.

The efficiency of the machine is quite good, compared to conventional single phase motors.

The motor can be run in either direction depending upon the initial position of the rotor with respect to the stater winding.

From the above discussion one can suggest some fields of application. The motor will be particularly suitable in the fractional and sub-fractional range, for applications involving light starting duty.

Boforo the motor can be a commercially successful one, the main two draw-backs should be over-comes

1) The motor can start only for some particular positions of the rotor with respect to stater winding. This domands for the positioning device to set the rotor in position. The author believes that it is also possible to construct such positioning device without such difficulty. Perhaps, the method suggested by Desai²⁹ may be adopted.

11) The motor has a poor starting and accolerating torque. BCOPH FOR FURTHER STUDY:

It may be noted that the forgoing results are for the experimental machine built from available materials and without experience on the design of this type of machine. The author believes it will be possible to improve the performance of the machine which needs a number of machines to be designed and studied. Also the experimental machine used a stamping which is normally meant for a four-pole design. The core was heavily saturated because of this and the no-load current is higher.

Some other problems which need further study are the offects of harmonics and their suppression. The stater current contains (1-20) and (3-20) frequency components in addition to the fundamental. Attempt may be made to reduce these harmonics. Further with the help of the digital computer, the various transients can be studied, and the machine parameters can also be optimised.

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APPENDIX-I

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REDUCTION OF MATRIX:

The work involved by the routine calculation in elimination of arrays of a matrix can be avoided and the reduced array can be written from the full array only by inspection of the following rules are applied:

1) The denominator of the fraction for each element consists of the determinant of those rows and columns which are to be eliminated which are also common to both.

2) The numerator of the fraction for each element consists of the determinant formed by the rows and columns to be eliminated which are common to the element being considered, together with the rows and columns of the dinominator.

This rule becomes very simple when only one row and columns are eliminated at a time, especially when several elements in the array are zero.

a ₁₁	°12	^a 13	814
^a 21	⁸ 22	⁸ 23	⁸ 24
⁸ 31	⁸ 32	⁸ 33	⁸ 34
⁸ 41	⁸ 42	⁸ 43	•44

If the last row and column are required to be eliminated from.

The result is-

a ₁₁ - <u>a₁₄ · c₄₁</u>	a ₁₂ - a ₁₄ . a ₄₂	•13- <u>•14 • • •43</u>
<u>a₄₄</u>	a ₄₄	•44
a ₂₁ - a ₂₄ • a ₄₁	22 ⁸ 24 ⁸ 42	•23" ⁸ 24 • ⁸ 43
a ₄₄	844	⁸ 44
^a 31 ⁻ ^a 34 • ^a 41	32 <u>834</u> 842	^a 33 ⁻ ^a 34 ^{• a} 43
^a 44	844	^a 44

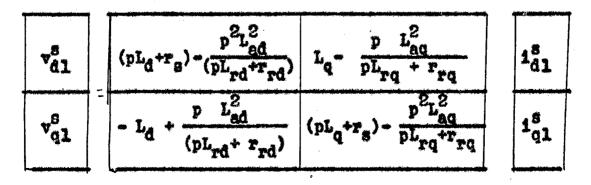
Applying the above inspection rule the matrix (1-5) can be reduced first to a 3 x 3 matrix and then to a 2 x 2 matrix,

Thus the result is-

By 1	lst	stage	of.	reduction
------	-----	-------	-----	-----------

v ^s d1		pL _d + r _s	$L_{q} = \frac{p L_{aq}^{2}}{p L_{rq} + r_{rq}}$	pLad	i ^s d1
vgl	*	- La	$(pL_q+r_s) = \frac{p^2L_{aq}^2}{pL_{rq}+r_{rq}}$	- L _{ad}	1 <mark>8</mark> 1 <u>8</u> 2
v _{q1}		- pLad		^{pL} rd ^{-r} rd	1 [#] 11

and by 2nd stage of reduction-



Now

Now putting $(L_1 + L_{ad}) = L_d \stackrel{*}{\leftarrow} (L_1 + L_{aq}) = L_q$

and
$$(L_{rl} + L_{ad}) = L_{rd} \triangleq (L_{rl} + L_{ad}) = L_{rd}$$

 $(pL_d + r_s) = \frac{p^2 L_{ad}^2}{(pL_{rd} + r_{rd})}$
 $= r_s + p (L_l + L_{ad}) = \frac{p^2 L_{ad}^2}{p(L_{rl} + L_{ad}) + r_{rd}}$
 $= r_s + \frac{p(L_l + L_{ad}) Ip(L_{rl} + L_{ad}) + r_{rd} - \frac{pL_{ad}^2}{(L_l + L_{ad}) + r_{rd} - \frac{pL_{ad}^2}{(L_l + L_{ad}) + r_{rd}}}{p(L_{rl} + L_{ad}) + r_{rd}}$
 $= r_s + \frac{I pL_d I \triangleq \frac{1}{r_{rd}} p(L_{rl} + \frac{L_l - L_{ad}}{L_l + \frac{L_{ad}}{L_l}})I$
 $= r_s + \frac{I pL_d I \triangleq \frac{1}{r_{rd}} p(L_{rl} + \frac{L_l - L_{ad}}{L_l + \frac{L_{ad}}{L_l}})I$
 $= r_s + pI_{id} \frac{1 + T_d p}{T_{do} p}$

$$= r_s + pL_d(p)$$

where,

$$T_{d}^{i} = \frac{1}{r_{rd}} (L_{r1} + \frac{L_{1} L_{ad}}{L_{1} + L_{ad}})$$

$$T_{\rm do} = \frac{1}{r_{\rm rd}} \left(L_{\rm rl} + L_{\rm ad} \right)$$

Again ,

$$L_{q} = \frac{pL_{aq}^{2}}{pL_{rq} + r_{rq}}$$

$$= (L_{1} + L_{aq}) - \frac{p^{2}L_{aq}^{2}}{p(L_{r1} + L_{aq}) + r_{rq}}$$

$$= \frac{(L_{1} + L_{aq})r_{rq} + p(L_{1} + L_{aq})L_{r1} + L_{aq}}{p(L_{r1} + L_{aq}) + r_{rq}}$$

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$$= \frac{(L_{1} + L_{aq}) I 1 + \frac{1}{r_{rq}} p (L_{g1} + \frac{L_{1} - L_{aq}}{L_{1} + L_{aq}})}{IR + \frac{1}{r_{rq}} p^{i} (L_{r1} + L_{aq}) I}$$

$$= L_{q} \frac{1 + T_{q}^{i}}{1 + T_{q0}^{i}}$$

where,

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= $L_q(p)$

$$T_q^* = \frac{1}{r_{rq}} (L_{r1} + \frac{L_1 L_{aq}}{L_1 + L_{aq}})$$

$$T_{qo}^{\dagger} = \frac{1}{T_{rq}} (L_{r1} + L_{aq})$$

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APPENDIX - II

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CALCULATION OF GAP CO-SFFICIENT:

In calculation of gap co-efficients the following assumptions are made:

1) The air-gap 'g' under the salist -pole is assumed to be uniform.

2) The air-gap ${}^{*}S_{0}$ under the inter-pole is also assumed to be uniform.

3) With uniform air-gap the spatial flux distribution in the air-gap is taken to be sinusoidal.

4) The flux enters the rotor radially even over the portions occupied by the inter-pole, but the magnitude is reduced in proportion.

With the assumptions made the flux density wave in the air-gap portion can be represented as in Fig.1.

 $B(\Theta) = B_1 \sin \Theta \qquad O \ \langle \Theta \ \langle \alpha \rangle$ $= B_2 \sin \Theta \qquad \alpha \ \langle \Theta \ \langle \pi/2 \rangle$

where,

$$B_1 = \frac{h_0 N}{s_0}$$
 and $B_2 = \frac{h_0 N}{s}$

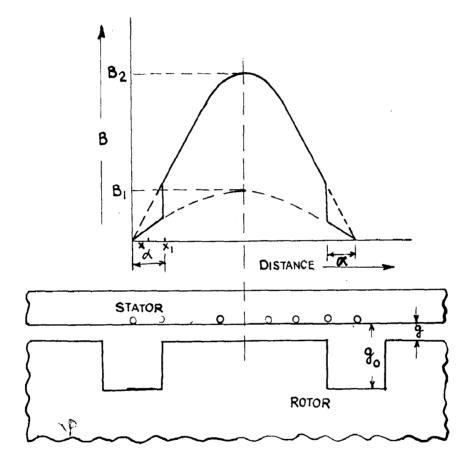
The fundamental component of the wave can be analysed by Fourier Series analysis.

$$b_{1} = \frac{4}{\pi} \left[\int_{0}^{N_{2}} B(\theta) \sin \theta d\theta \right]$$

$$= \frac{4}{\pi} \left[\int_{0}^{A} B_{1} \sin^{2} \theta d\theta + \int_{A}^{N_{2}} B_{2} \sin^{2} \theta d\theta \right]$$

$$= \frac{4}{\pi} / \int_{0}^{A} M \left[\int_{0}^{A} \frac{1}{\xi_{0}} \sin^{2} \theta d\theta + \int_{A}^{N_{2}} \frac{1}{\xi} \sin^{2} \theta d\theta \right]$$

$$= \frac{4}{\pi} / \int_{0}^{A} M x + \left[\frac{1}{\xi_{0}} \int_{0}^{A} (1 - \cos 2\theta) d\theta + \frac{1}{\xi} \int_{A}^{N_{2}} (3 - \cos 2\theta) d\theta \right]$$



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$$= \frac{2}{7} \mathcal{K}_{OM} \left[\frac{1}{5} \left(\alpha - \frac{1}{2} \sin 2\alpha \right) + \frac{1}{2} \left[\left(\frac{\pi}{2} - \alpha \right) + \frac{1}{2} \sin 2\alpha \right] \right]$$

.*. The peak flux-density at any point on the salient polar region

$$B_{d} = \frac{2}{\pi} h_{0} N \left[\frac{1}{80} (\alpha - \frac{1}{8} \sin 2\alpha) + \frac{1}{8} I (\frac{\pi}{2} - \alpha) + \frac{1}{8} \sin 2\alpha \right]$$

.*. To calculate Cat

$$C_{d} = \frac{\text{Smooth rotor flux}}{\text{Direct axis flux}} = \frac{\beta_{n}}{\beta_{d}} = \frac{B_{n}}{B_{d}}$$
Now, $B_{n} = \frac{4}{\pi} \int_{0}^{\frac{\pi}{2}} B(\theta) \sin \theta \, d\theta$

$$= \frac{4}{\pi} \int_{0}^{\frac{\pi}{2}} \frac{\hbar}{\sqrt{\theta}} \frac{M}{8} \sin^{2} \theta \, d\theta$$

$$= \frac{2}{\pi} \int_{0}^{\frac{\pi}{2}} \frac{\hbar}{\sqrt{\theta}} \frac{M}{8} \pi \frac{1}{8} \int_{0}^{\frac{\pi}{2}} (1 - \cos 2\theta) \, d\theta$$

$$= \frac{1}{\pi} \int_{0}^{\frac{\pi}{2}} (\alpha - \frac{1}{8} \sin 2\alpha) + \frac{1}{8} - \frac{1}{8} (\pi/2 - \alpha) + \frac{1}{8} \sin 2\alpha]$$

To calculate Cg:

The peak flux-density an any point on the inter-polar region:

$$B_{q} = \frac{4}{\pi} \left[\int_{0}^{\frac{1}{2}-\alpha} \dot{B}_{2} \sin^{2}\theta d\theta + \int_{\frac{\pi}{2}-\alpha}^{\frac{\pi}{2}} B_{1} \sin^{2}\theta d\theta \right]$$

$$= \frac{4}{\pi} \int_{0}^{\pi} M \left[\frac{1}{4} \int_{0}^{\frac{\pi}{2}-\alpha} \sin^{2}\theta d\theta + \frac{1}{4} \int_{0}^{\frac{\pi}{2}} \sin^{2}\theta d\theta \right]$$

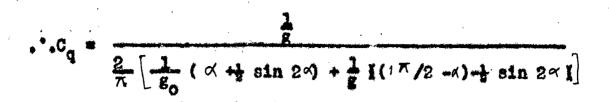
$$= \frac{4}{\pi} \int_{0}^{\pi} M x_{2}^{4} \left[-\frac{1}{4} \int_{0}^{\frac{\pi}{2}-\alpha} (1 - \cos 2\theta) d\theta + \frac{1}{4} \int_{0}^{\frac{\pi}{2}-\alpha} (1 - \cos 2\theta) d\theta \right]$$

$$= \frac{2}{\pi} \int_{0}^{\pi} M \left[-\frac{1}{4} \right] \left[(\frac{\pi}{2} - \alpha) + \sin(\pi - 2\alpha) \right] + \frac{1}{4} \left[(\alpha + \frac{1}{4} \sin 2\alpha) \right]$$

$$= \frac{2}{\pi} \int_{0}^{\pi} M \left[-\frac{1}{4} \right] \left[(\frac{\pi}{2} - \alpha) + \frac{1}{4} \sin(\pi - 2\alpha) \right] + \frac{1}{4} \left[(\alpha + \frac{1}{4} \sin 2\alpha) \right]$$

$$= \frac{3\pi}{4} \int_{0}^{\pi} M \left[-\frac{1}{4} \left[(\frac{\pi}{2} - \alpha) + \frac{1}{4} \sin(\pi - 2\alpha) \right] + \frac{1}{4} \int_{0}^{\pi} (\alpha + \frac{1}{4} \sin 2\alpha) \right]$$

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Now, putting the values from fig.(A1), we can calculate-

 $C_d = 1.04$ when $2d = 49^{\circ}5$

 and
 $C_q = 1.90$ g = 0.016 mch.

 $g_{\circ} = 0.312$ in
 $g_{\circ} = 0.312$ in

APPENDIX-III

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COMPUTER PROGRAMMING FOR TRANSIENT STUDY:

C C BOLN OF DIFF BQN FOR TRANS COND TK CHATTERJEE EED UOR DIMENSION 2(6), PD(6),Q(6),F(5,5) DIMENSION AX(4), CX(4),H(6), EX(4)

PUNCH 10

10 FORMAT (SHTHETA, 12X, 2HWR, 10X, 2HI1, 10X, 3HID2, 9K, 3HIQ2, 9X, 4HTIME) PI#22.0/7.0

W #2.0*PI*50.0

TLOO

TB=0.

VI= 230.0+1.414

PU#3.5*(10.**(-3))

RI=12.1

RD2=10.5

RQ2=8.0

AL11#1.38

ALD2=1.42

ALQ2=1.41

HMD=520.0/W

HMQ=370.0/W

READL, THETA

```
1 FORMET(P10.2)
```

DO 2 1=1,6

Z(I)=0.0

PD(1)=0.0

Q(I)=0.0

S CONTINUE

\$(1)=THETE*PI/180.0

T=0.0

TS=0.0005

BD(6)=1.0

AN=1.

HMT=1,0+Z(1)

F(1,1)=AL11

F(2,2)=ALD2

F(3.3)=ALQ2

5 F(1,2)=HMD*COSF(HMT)

F(2,1)=F(1,2)

F(1,3)=>HNQ*SINF(HMT)

F(3,1)=F(1,3)

F(2,3)=0.0

F(3.2)=0.0

F=V1+SINF(W+T)+F(1,3)+2(2)+2(4)

F(1,4)=FF-F(1,2)*Z(2)*Z(5)-R1*Z(3)

F(2,4)=F(1,3)*Z(2)*Z(3)-RD2*Z(4)

F(3,4) =F(1,2) +Z(2) +Z(3) -RQ2+Z(5)

CALL SOLEQN(F)(3,4))

DO 3 I=1,3

```
3 PD(1+2)=P(1,4)
```

PD(1)=2(2)

TE=-2(3) + 2(4) + F(1,3) + 2(3) + 2(6) + F(1,2)

PD(2)=TE/PJ

AX[1)=0.5

AX(2)=1.0-SQRTF(0.5)

AX(3)=1.0+SQRTF(0,5)

AX(4)=1.0/6.0

BX(1)=2.0

BX(2)=1.0

BX(3)=1.0

BX(4)=2.0

CX(1)=0.5

CX(2)=AX(2)

CX(3)=AX(3)

CX(4)=0.5

02 4 I=1,6

DO 4 J=1,4

H(I)=TS*PD(I)

R=AX(J)+DH(I)-BX(J)+Q(I))

Z(I)=Z(I)+R

Q(I)=Q(I)+3.0+R-CX(J)+H(I)

4 CONTINUE

PUNCH6, (Z(I), I=1,6)

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6 FORMAT(6E12.6)

60 TO 8

STOP

END

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APPENDIX-IV

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DESIGN CONSIDERATION:

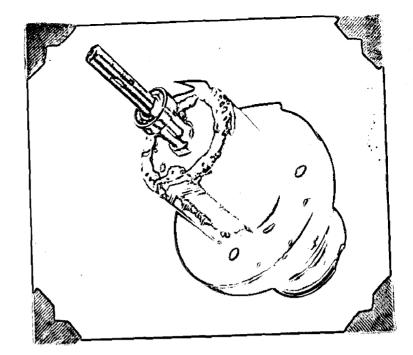
The problem of design of any rotating machine can be considered in two parts. The first part to calculate the performance of the machine from the knowledge of the machine geometry and winding details. This is normally done by calculating certain basic machine constants and then applying the methods of machine analysis to calculate the performance. The second part of the design problem is to determine the machine geometry and winding details from the performance specifications. This has to be more or less always tackled by trial and error method. Since a large number of solutions are possible in this case, a certain amount of experience is necessary to arrive at a satisfactory solution without spending excessive amount of time.

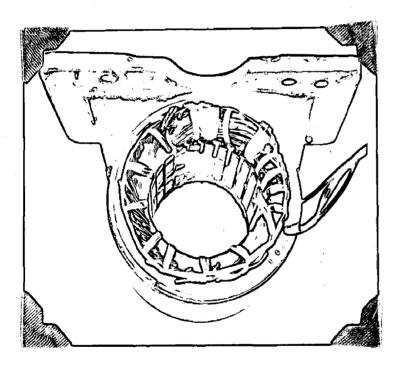
It may be added that this machine has not been commercially manufactured uptil now and hence large amount of experimental data is not available to verify the methods presented here. This study is presented here simply as a begining and many assumptions, formulae etc. can be modified later in the light of experimental results. In fact, the experimental method of determining machine constants itself can be improved.

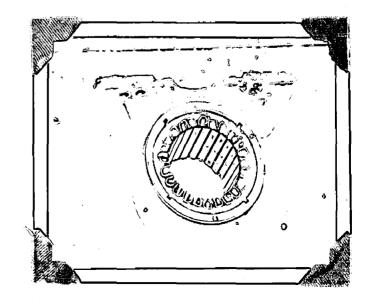
The following procedure as followed by the author may be suggested tentatively, but will be modified in light of further experience. Familiarity with design of normal types of single phase motors is assumed.

- 1) The frame size and main dimensions can be kept same as fer a split-phase machine.
- 2) The stator stamping can be selected as for a normal motor. Since no auxiliary winding is needed, it would be

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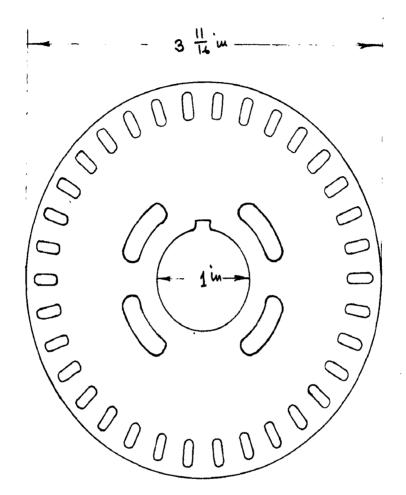


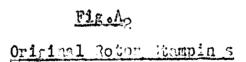
desirable to omit slots normally used for auxiliary winding and fill the slot spaces.

- 3) The stator winding can be designed as for normal motors to secure a good m.m.f. wave shape.
- 4) The air-gap should be kept as large as possible to reduce undesirable effects of harmonics.
- 5) The rotor stamping can be selected as for normal motors. The end ring section can be selected to give about 10 to 15 percent of rotor resistance. Closed slots are desirable in reducing troubles due to permeance harmonics, although it will contribute to higher leakage.
- 6) Skewing is a considered absolutoly essential.
- The amount of rotor dissymmetry shall be kept as low as
 percent or so.

The design can be completed following the usual trial and error method, to achieve the desired performance.

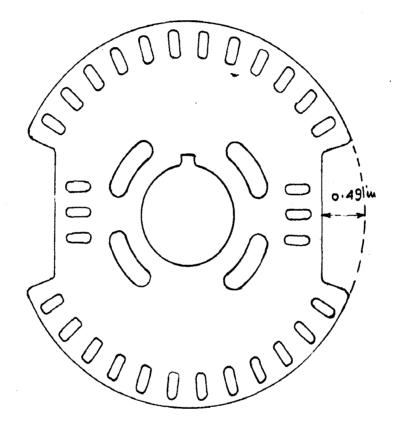
1 h.b. 230 single phase, 5045 Principal Design datas Stator framot No. of slots a 24 Tooth pitch = 23 in - 53 in Pole pitch = 6 in. Outside dia = 3 in, Boro dia Slack longth = 3 in. Material - Special Lohys (0.020" thick) steel manufactured by M/s. Sankoy Electrical Stampings Ltd.

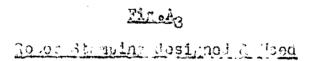




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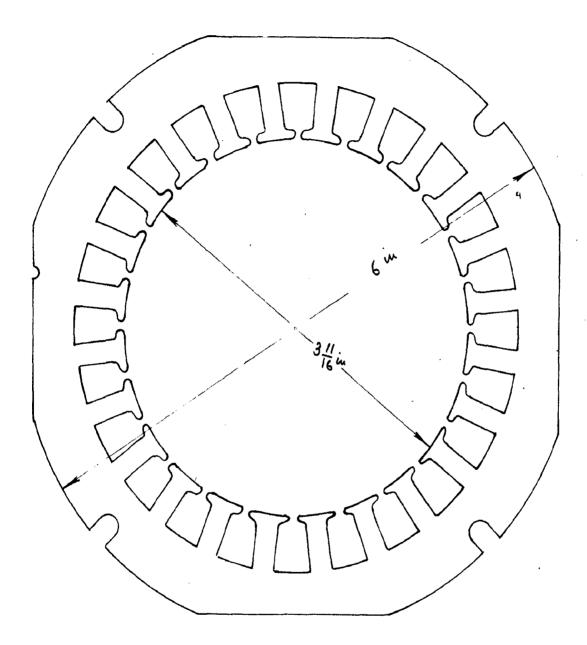


Fig.A.

. Stator Stantar user

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Stator Windings

No. of colls	≈284
frequency	= 50 c/s.
Coil pitch	B D
Ro.of poles	= 2
No.of turns por	· coll = 22
No.of turns in	sories= 528
_	strands of 22 S.W.G. super enamelled
coppor wire.	
Double layer ro	sistance at $25^{\circ}C = 12.1$ ohns.
Stator leakage	roactanco = 15.4 ohns.

Rotor Stamping:

Out dia	= 3 11 in. (unfinished)
Alr-gop	= 0.016 in.
Boro dia	= 1 in
No.of slots	(unshaped) = 36
Ro.of slots	(After shaping) = 30
Stack longth	$1 = 2 - \frac{63}{60} 1n.$
Matorial - S	Special Lohy's (0.020" thick) steel
П	nanufacturod by M/s. Sankey Electrical
2	Stampings Ltd.

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